

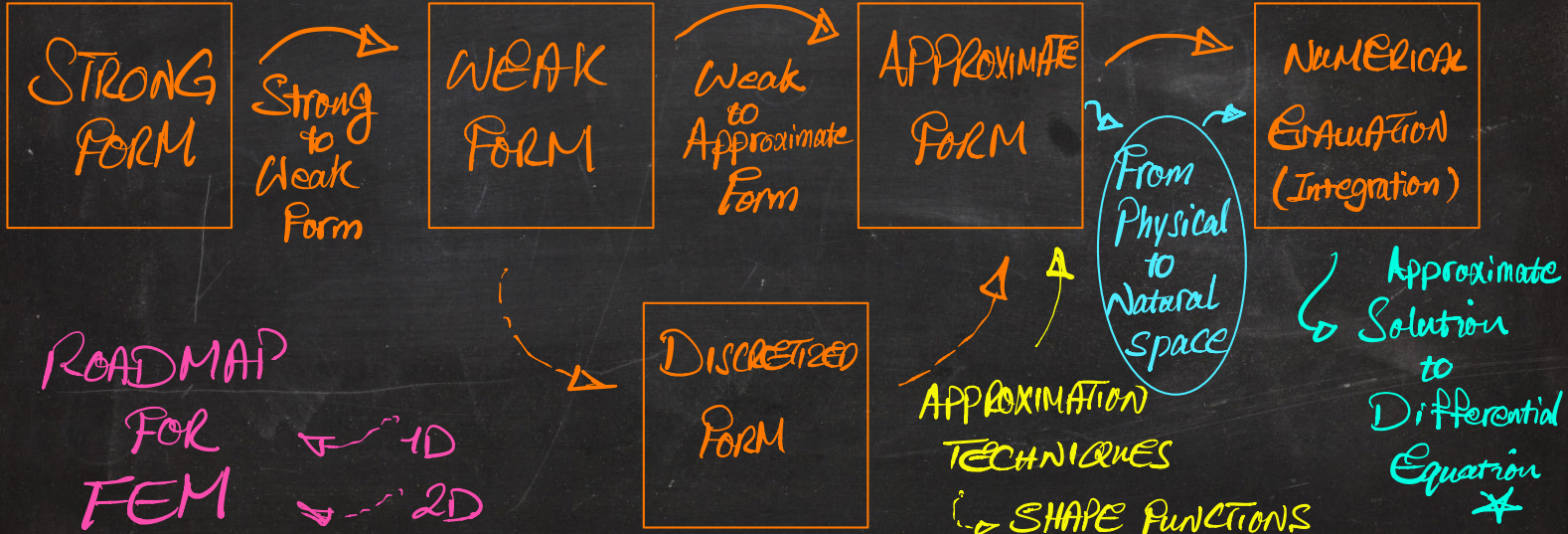
FINITE ELEMENT METHOD

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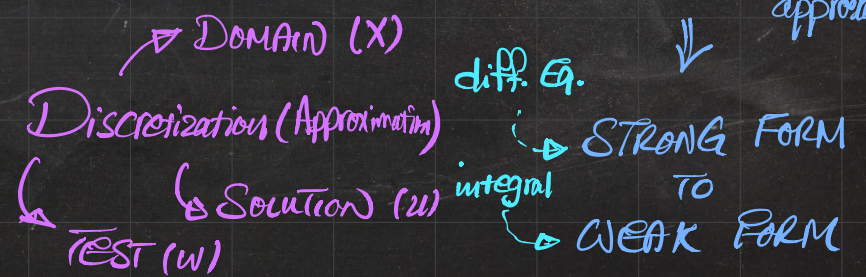
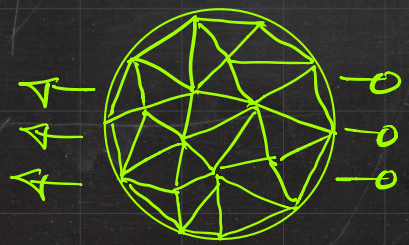
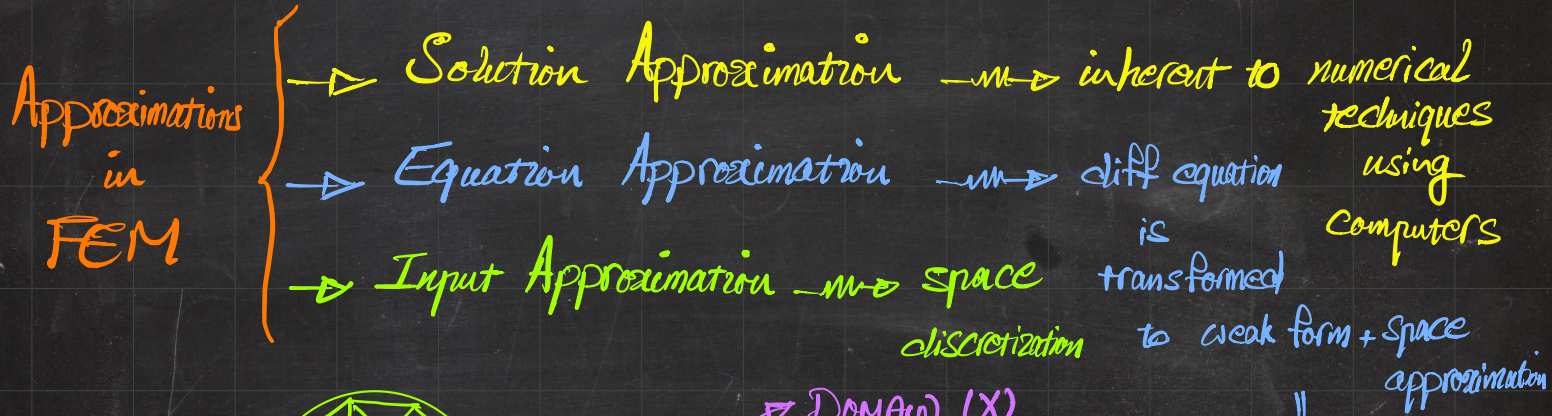
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FINITE ELEMENT METHOD

Differential Equation \star



UNDERSTANDING FEM VIA AN ANALOGY (A BRUTAL SIMPLIFICATION)



1D FEM

Overview and Wrap-up

FINITE ELEMENT METHOD IN A NUTSHELL

Diff. Eq. $(EAu')' + b = 0$
 2ND. O.D.E.

STRONG FORM

(I) MULTIPLY BY w (test function)
 (II) INTEGRATE

WEAK FORM

INTEGRAL FORM

$$\int_0^1 w'u' dx = \int_0^1 w da + w(1)u'(1) - w(0)u'(0)$$

PIECEWISE

APPROXIMATE FORM

Approximate Discretized Weak Form

Approximation

DISCRETIZED FORM

NUMERICAL INTEGRATION
 another source of approx...

ELEMENT-WISE QUANTITIES

SOLVE

PostProcess

GLOBAL SYSTEM

$$[K][u] = [F]$$

FROM GLOBAL TO ELEMENTS

FROM INTEGRAL OVER THE DOMAIN TO SUBINTEGRALS

$$\int_0^1 \dots dx = \int_0^a \dots dx + \int_a^b \dots dx + \dots$$

PIECEWISE INTEGRALS (SOLUTIONS)

ASSEMBLY

FROM STRONG FORM TO ELEMENT STIFFNESS IN PHYSICAL SPACE

$EA u'' = 0$ SUBJECT TO BCs



MULTIPLY BY
TEST
FUNCTION
 w

$\left\{ \begin{array}{l} \text{DIRICHLET} \rightarrow u \text{ IS PRESCRIBED} \\ \text{NEUMANN} \rightarrow u' \text{ IS PRESCRIBED} \end{array} \right.$

$EA w u'' = 0 \rightarrow w u'' = (w u')' - w' u'$

$EA [(w u')' - w' u'] = 0 \Rightarrow EA w' u' = EA (w u')'$ INTEGRATE

$\int_L EA w' u' dx = \int_L EA (w u')' dx = EA w u' \Big|_1^2 = EA w^2 u'^2 - EA w^1 u'^1$

FROM STRONG FORM TO ELEMENT STIFFNESS \rightarrow IN PHYSICAL SPACE NO.23

$$EA u'' = 0 \quad \text{SUBJECT TO BCs}$$



$$EA \begin{bmatrix} \int_L N^1{}' N^1{}' dx & \int_L N^1{}' N^2{}' dx \\ \int_L N^2{}' N^1{}' dx & \int_L N^2{}' N^2{}' dx \end{bmatrix} \begin{bmatrix} u^1 \\ u^2 \end{bmatrix} = \begin{bmatrix} F^1 \\ F^2 \end{bmatrix}$$

$$\begin{bmatrix} K^{11} & K^{12} \\ K^{21} & K^{22} \end{bmatrix} \begin{bmatrix} u^1 \\ u^2 \end{bmatrix} = \begin{bmatrix} F^1 \\ F^2 \end{bmatrix} \quad \rightarrow \quad K^{ij} = EA \int_L N^i{}' N^j{}' dx$$

$$K^{ij} = EA \int_L n^i n^j dx$$

PHYSICAL RECALL:

$$= EA \int_{-1}^1 \frac{\partial n^i}{\partial \xi} \frac{\partial n^j}{\partial \xi} J^{-1} d\xi$$

NATURAL

$$\int_{-1}^1 g(\xi) d\xi = \sum_{gp=1}^{GPE} g(\xi) \alpha_{gp}$$

Loop over gp

$$= EA \sum_{gp=1}^{GPE} \left\{ \left[\frac{\partial n^i}{\partial \xi} \quad \frac{\partial n^j}{\partial \xi} \quad J^{-1} \right] \times \alpha_{gp} \right\}$$

END

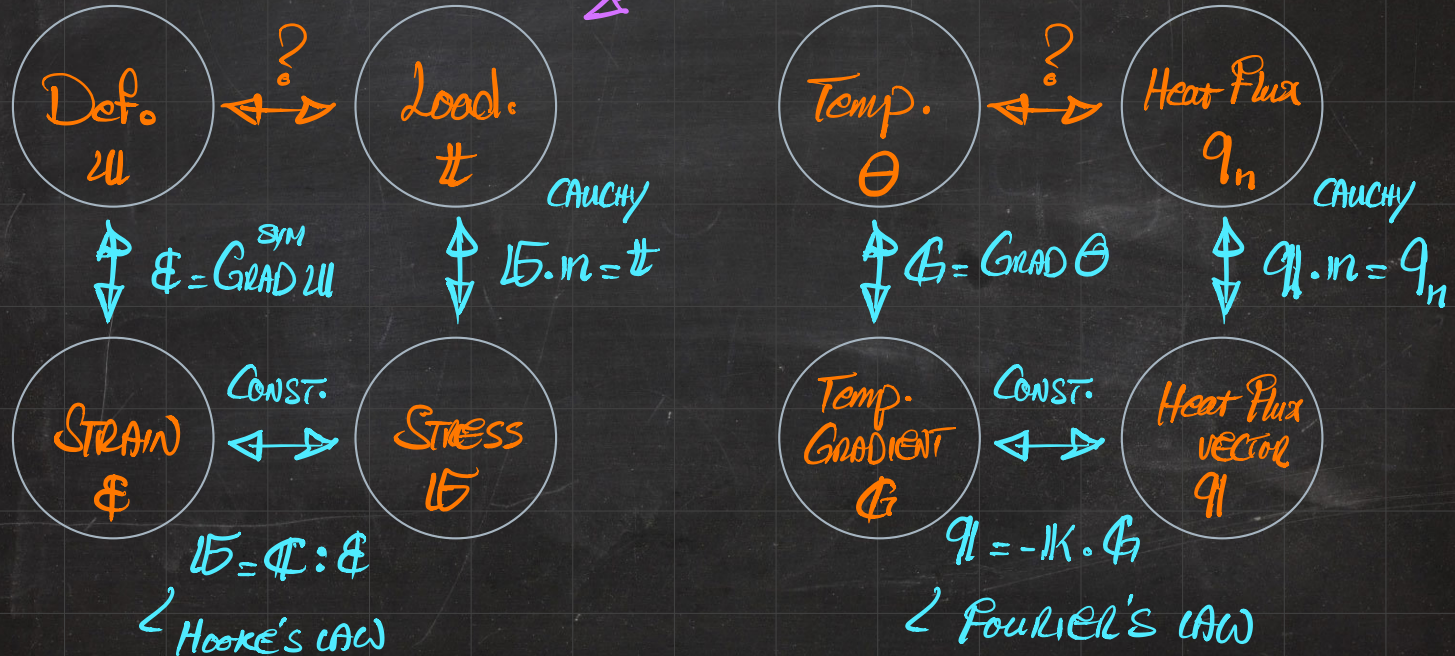
WHAT YOU SEE IN THE CODE!

For gp=1:GPE
 ...
 End

eg. in MATLAB

2D FEM

Big Picture of Mechanics (Mechanical Problems & Thermal Problems)



STRONG FORM (GENERIC FORM) $\rightarrow \text{Div } \mathbf{B} + \mathbf{b} = 0$, $\text{Div } \mathbf{q} + c = 0$

$$\frac{\partial B_{jk}}{\partial x_k} + b_j = 0$$

$$\frac{\partial q_i}{\partial x_i} + c = 0$$

$$\left\{ \begin{array}{l} \frac{\partial B_{xx}}{\partial x} + \frac{\partial B_{xy}}{\partial y} + b_x = 0 \\ \frac{\partial B_{yx}}{\partial x} + \frac{\partial B_{yy}}{\partial y} + b_y = 0 \end{array} \right.$$

$$\frac{\partial q_x}{\partial x} + \frac{\partial q_y}{\partial y} + c = 0$$

2D \rightarrow Plane STRAIN } $\mathbf{B} = \mathbf{F} \cdot \mathbf{F}$
 Plane STRESS }

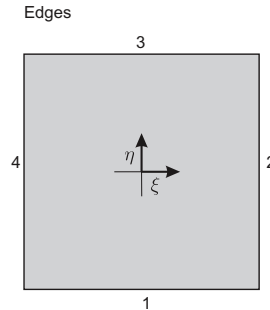
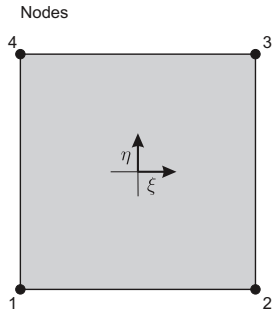


- two-dimensional 4-noded quadrilateral element (D2QU4N)
a.k.a. bilinear quadrilateral element
- two-dimensional 9-noded quadrilateral element (D2QU9N)
a.k.a. Lagrange biquadratic quadrilateral element
- two-dimensional 8-noded quadrilateral element (D2QU8N)
a.k.a. serendipity biquadratic quadrilateral element
- two-dimensional 3-noded triangular element (D2TR3N)
a.k.a. constant strain triangle
- two-dimensional 6-noded triangular element (D2TR6N)
a.k.a. quadratic triangle
- two-dimensional quadrature rule

2D Finite Element Library

D2QU4N

bilinear quadrilateral element



Node Number	Coordinates	
	ξ	η
1	-1	-1
2	1	-1
3	1	1
4	-1	1

$$N^1 = \frac{1}{4} (1 - \xi) (1 - \eta)$$

$$N^2 = \frac{1}{4} (1 + \xi) (1 - \eta)$$

$$N^3 = \frac{1}{4} (1 + \xi) (1 + \eta)$$

$$N^4 = \frac{1}{4} (1 - \xi) (1 + \eta)$$

$$N_{,\xi}^1 = -\frac{1}{4} (1 - \eta)$$

$$N_{,\xi}^2 = +\frac{1}{4} (1 - \eta)$$

$$N_{,\xi}^3 = +\frac{1}{4} (1 + \eta)$$

$$N_{,\xi}^4 = -\frac{1}{4} (1 + \eta)$$

$$N_{,\eta}^1 = -\frac{1}{4} (1 - \xi)$$

$$N_{,\eta}^2 = -\frac{1}{4} (1 + \xi)$$

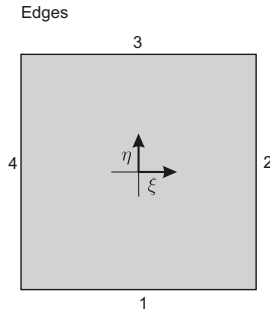
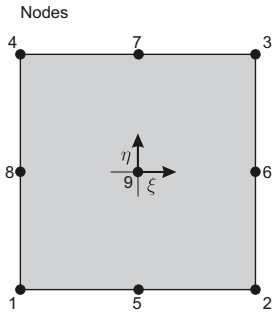
$$N_{,\eta}^3 = +\frac{1}{4} (1 + \xi)$$

$$N_{,\eta}^4 = +\frac{1}{4} (1 - \xi)$$

2D Finite Element Library

D2QU9N

Lagrange biquadratic quadrilateral element



Node Number	Coordinates	
	ξ	η
1	-1	-1
2	1	-1
3	1	1
4	-1	1
5	0	-1
6	1	0
7	0	1
8	-1	0
9	0	0

$$N^1 = +\frac{1}{4} (1 - \xi) \xi (1 - \eta) \eta$$

$$N^2 = -\frac{1}{4} (1 + \xi) \xi (1 - \eta) \eta$$

$$N^3 = +\frac{1}{4} (1 + \xi) \xi (1 + \eta) \eta$$

$$N^4 = -\frac{1}{4} (1 - \xi) \xi (1 + \eta) \eta$$

$$N^5 = -\frac{1}{2} (1 - \xi) (1 + \xi) (1 - \eta) \eta$$

$$N^6 = +\frac{1}{2} (1 + \xi) \xi (1 - \eta) (1 + \eta)$$

$$N^7 = +\frac{1}{2} (1 - \xi) (1 + \xi) (1 + \eta) \eta$$

$$N^8 = -\frac{1}{2} (1 - \xi) \xi (1 - \eta) (1 + \eta)$$

$$N^9 = (1 - \xi) (1 + \xi) (1 - \eta) (1 + \eta)$$

$$N_{,\xi}^1 = +\frac{1}{4} (1 - 2\xi) (1 - \eta) \eta$$

$$N_{,\xi}^2 = -\frac{1}{4} (1 + 2\xi) (1 - \eta) \eta$$

$$N_{,\xi}^3 = +\frac{1}{4} (1 + 2\xi) (1 + \eta) \eta$$

$$N_{,\xi}^4 = -\frac{1}{4} (1 - 2\xi) (1 + \eta) \eta$$

$$N_{,\xi}^5 = \xi \eta (1 - \eta)$$

$$N_{,\xi}^6 = \frac{1}{2} (1 + 2\xi) (1 - \eta) (1 + \eta)$$

$$N_{,\xi}^7 = -\xi \eta (1 + \eta)$$

$$N_{,\xi}^8 = -\frac{1}{2} (1 - 2\xi) (1 - \eta) (1 + \eta)$$

$$N_{,\xi}^9 = -2\xi (1 - \eta) (1 + \eta)$$

$$N_{,\eta}^1 = +\frac{1}{4} (1 - \xi) \xi (1 - 2\eta)$$

$$N_{,\eta}^2 = -\frac{1}{4} (1 + \xi) \xi (1 - 2\eta)$$

$$N_{,\eta}^3 = +\frac{1}{4} (1 + \xi) \xi (1 + 2\eta)$$

$$N_{,\eta}^4 = -\frac{1}{4} (1 - \xi) \xi (1 + 2\eta)$$

$$N_{,\eta}^5 = \frac{1}{2} (1 - \xi) (1 + \xi) (2\eta - 1)$$

$$N_{,\eta}^6 = -(1 + \xi) \xi \eta$$

$$N_{,\eta}^7 = \frac{1}{2} (1 - \xi) (1 + \xi) (1 + 2\eta)$$

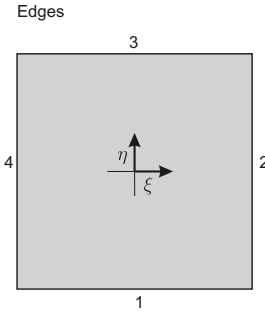
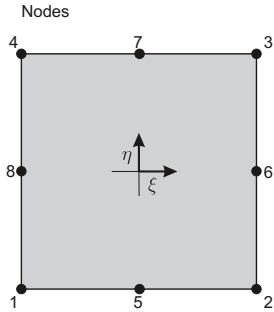
$$N_{,\eta}^8 = (1 - \xi) \xi \eta$$

$$N_{,\eta}^9 = -2(1 - \xi) (1 + \xi) \eta$$

2D Finite Element Library

D2QU8N

serendipity biquadratic quadrilateral element



Node Number	Coordinates	
	ξ	η
1	-1	-1
2	1	-1
3	1	1
4	-1	1
5	0	-1
6	1	0
7	0	1
8	-1	0

$$N^1 = -\frac{1}{4} (1 - \xi) (1 - \eta) (1 + \xi + \eta)$$

$$N^2 = -\frac{1}{4} (1 + \xi) (1 - \eta) (1 - \xi + \eta)$$

$$N^3 = -\frac{1}{4} (1 + \xi) (1 + \eta) (1 - \xi - \eta)$$

$$N^4 = -\frac{1}{4} (1 - \xi) (1 + \eta) (1 + \xi - \eta)$$

$$N^5 = \frac{1}{2} (1 - \xi) (1 + \xi) (1 - \eta)$$

$$N^6 = \frac{1}{2} (1 + \xi) (1 + \eta) (1 - \eta)$$

$$N^7 = \frac{1}{2} (1 - \xi) (1 + \xi) (1 + \eta)$$

$$N^8 = \frac{1}{2} (1 - \xi) (1 + \eta) (1 - \eta)$$

$$N_{,\xi}^1 = +\frac{1}{4} (1 - \eta) (2\xi + \eta)$$

$$N_{,\xi}^2 = +\frac{1}{4} (1 - \eta) (2\xi - \eta)$$

$$N_{,\xi}^3 = +\frac{1}{4} (1 + \eta) (2\xi + \eta)$$

$$N_{,\xi}^4 = +\frac{1}{4} (1 + \eta) (2\xi - \eta)$$

$$N_{,\xi}^5 = -\xi (1 - \eta)$$

$$N_{,\xi}^6 = +\frac{1}{2} (1 - \eta) (1 + \eta)$$

$$N_{,\xi}^7 = -\xi (1 + \eta)$$

$$N_{,\xi}^8 = -\frac{1}{2} (1 - \eta) (1 + \eta)$$

$$N_{,\eta}^1 = +\frac{1}{4} (1 - \xi) (\xi + 2\eta)$$

$$N_{,\eta}^2 = +\frac{1}{4} (1 + \xi) (-\xi + 2\eta)$$

$$N_{,\eta}^3 = +\frac{1}{4} (1 + \xi) (\xi + 2\eta)$$

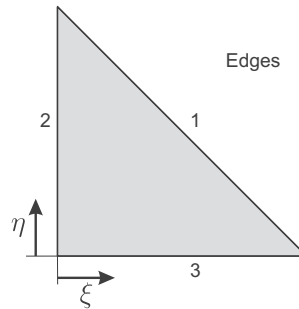
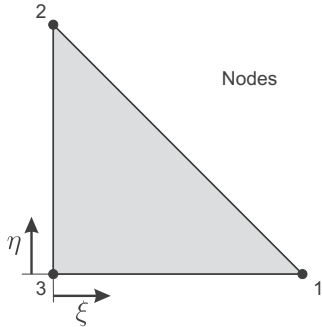
$$N_{,\eta}^4 = +\frac{1}{4} (1 - \xi) (-\xi + 2\eta)$$

$$N_{,\eta}^5 = -\frac{1}{2} (1 - \xi) (1 + \xi)$$

$$N_{,\eta}^6 = -(1 + \xi) \eta$$

$$N_{,\eta}^7 = +\frac{1}{2} (1 - \xi) (1 + \xi)$$

$$N_{,\eta}^8 = -(1 - \xi) \eta$$



Node Number	Coordinates	
	ξ	η
1	1	0
2	0	1
3	0	0

$$N^1 = \xi$$

$$N^2 = \eta$$

$$N^3 = (1 - \xi - \eta)$$

$$N_{,\xi}^1 = 1$$

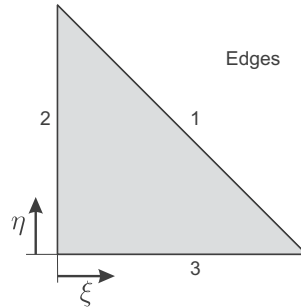
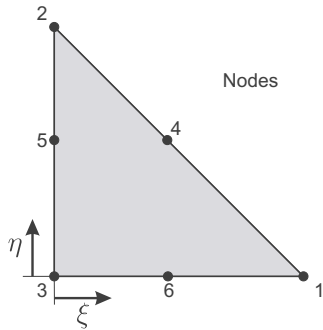
$$N_{,\xi}^2 = 0$$

$$N_{,\xi}^3 = -1$$

$$N_{,\eta}^1 = 0$$

$$N_{,\eta}^2 = 1$$

$$N_{,\eta}^3(\xi, \eta) = -1$$



Node Number	Coordinates	
	ξ	η
1	1	0
2	0	1
3	0	0
4	1/2	1/2
5	0	1/2
6	1/2	0

$$N^1 = \xi(2\xi - 1)$$

$$N_{,\xi}^1 = -1 + 4\xi$$

$$N_{,\eta}^1 = 0$$

$$N^2 = \eta(2\eta - 1)$$

$$N_{,\xi}^2 = 0$$

$$N_{,\eta}^2 = -1 + 4\eta$$

$$N^3 = (1 - \xi - \eta)(1 - 2\xi - 2\eta)$$

$$N_{,\xi}^3 = -3 + 4\xi + 4\eta$$

$$N_{,\eta}^3 = -3 + 4\xi + 4\eta$$

$$N^4 = 4\xi\eta$$

$$N_{,\xi}^4 = 4\eta$$

$$N_{,\eta}^4 = 4\xi$$

$$N^5 = 4\eta(1 - \xi - \eta)$$

$$N_{,\xi}^5 = -4\eta$$

$$N_{,\eta}^5 = -4(-1 + 2\eta + \xi)$$

$$N^6 = 4\xi(1 - \xi - \eta)$$

$$N_{,\xi}^6 = -4(-1 + \eta + 2\xi)$$

$$N_{,\eta}^6 = -4\xi$$

two-dimensional quadrature rule i

Triangular Elements Gauss Point Rule

$$\int_0^1 \int_0^{1-\eta} \{\bullet\} d\xi d\eta \approx \frac{1}{2} \sum_{i=1}^{N_{\text{Gauss Points}}} \alpha_i \{\bullet\}_{\text{Gauss Point } i}$$

Gauss Point Number	Coordinates		Weight Factor
	ξ	η	α
1	1/3	1/3	1

Gauss Point Number	Coordinates		Weight Factor
	ξ	η	α
1	1/6	1/6	1/3
2	4/6	1/6	1/3
3	1/6	4/6	1/3

two-dimensional quadrature rule ii

Quadrilateral Elements Gauss Point Rule

$$\int_{-1}^1 \int_{-1}^1 \{\bullet\} d\xi d\eta \approx \sum_{i=1}^{N_{\text{Gauss Points}}} \alpha_i \{\bullet\}_{\text{Gauss Point } i}$$

Gauss Point Number	Coordinates		Weight Factor
	ξ	η	α
1	0	0	2×2

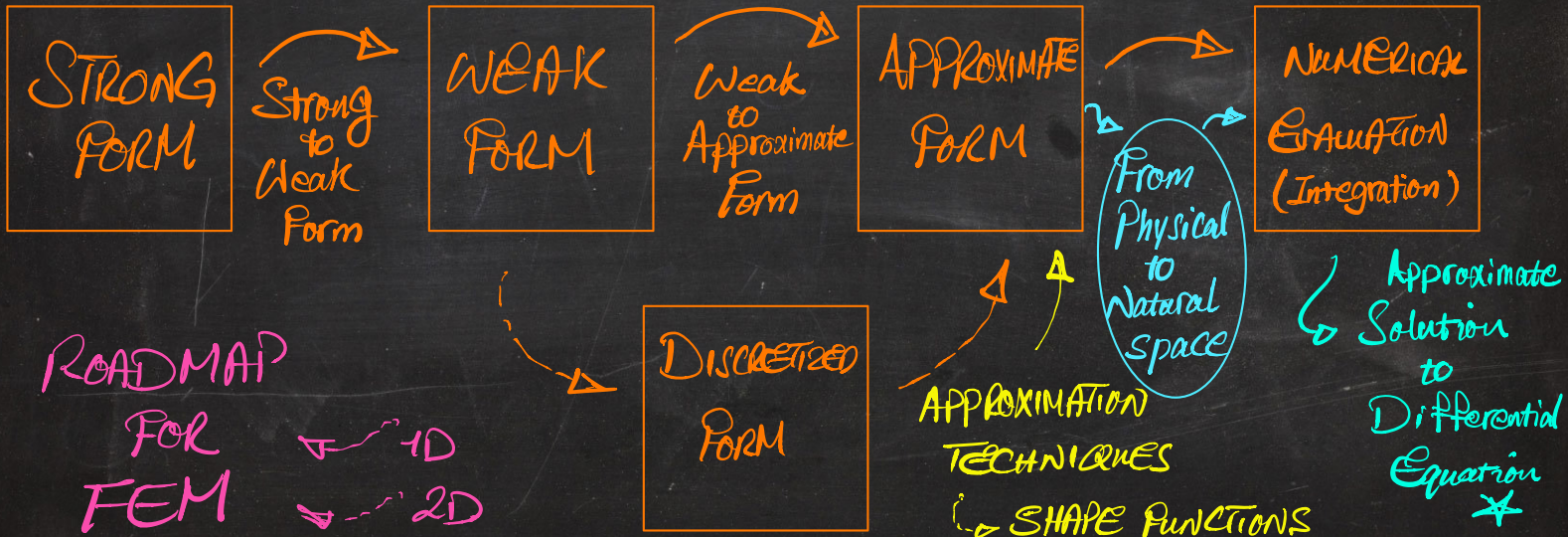
Gauss Point Number	Coordinates		Weight Factor
	ξ	η	α
1	$-1/\sqrt{3}$	$-1/\sqrt{3}$	1×1
2	$+1/\sqrt{3}$	$-1/\sqrt{3}$	1×1
3	$+1/\sqrt{3}$	$+1/\sqrt{3}$	1×1
4	$-1/\sqrt{3}$	$+1/\sqrt{3}$	1×1

two-dimensional quadrature rule iii

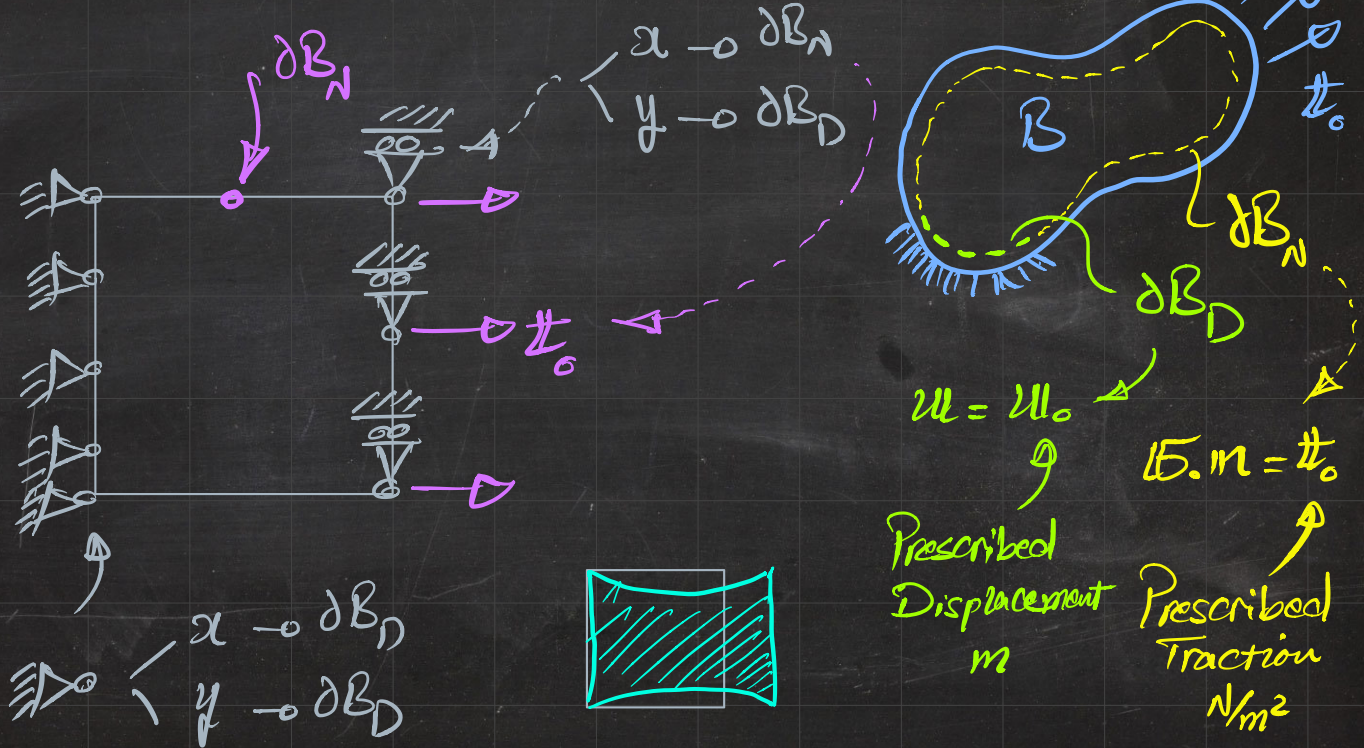
Gauss Point Number	Coordinates		Weight Factor
	ξ	η	α
1	$-\sqrt{3/5}$	$-\sqrt{3/5}$	$5/9 \times 5/9$
2	$+\sqrt{3/5}$	$-\sqrt{3/5}$	$5/9 \times 5/9$
3	$\sqrt{3/5}$	$\sqrt{3/5}$	$5/9 \times 5/9$
4	$-\sqrt{3/5}$	$\sqrt{3/5}$	$5/9 \times 5/9$
5	0	$-\sqrt{3/5}$	$5/9 \times 8/9$
6	$+\sqrt{3/5}$	0	$5/9 \times 8/9$
7	0	$\sqrt{3/5}$	$5/9 \times 8/9$
8	$-\sqrt{3/5}$	0	$5/9 \times 8/9$
9	0	0	$8/9 \times 8/9$

FINITE ELEMENT METHOD

Differential Equation *

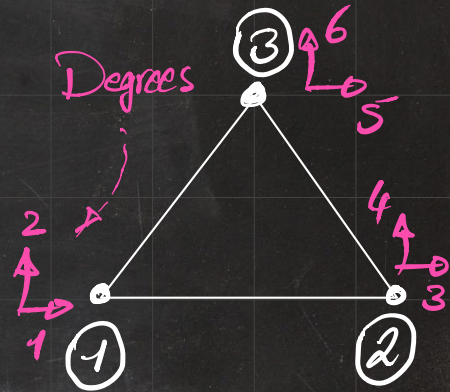


FROM STRONG FORM TO WEAK FORM



$$K_{\Delta} = \begin{bmatrix} K^{11} & & & & & \\ & K^{12} & & & & \\ & & K^{13} & & & \\ & K^{21} & & K^{22} & & K^{23} \\ & & & & K^{32} & & K^{33} \\ & K^{31} & & & & & & K^{33} \end{bmatrix}$$

6x6
Non-symmetric PD
↑₃ ↑₂



$$K_{\Delta} = \begin{bmatrix} \begin{matrix} K_{11}^{11} & K_{12}^{11} \\ K_{21}^{11} & K_{22}^{11} \end{matrix} & \begin{matrix} K_{11}^{12} & K_{12}^{12} \\ K_{21}^{12} & K_{22}^{12} \end{matrix} & \begin{matrix} K_{11}^{13} & K_{12}^{13} \\ K_{21}^{13} & K_{22}^{13} \end{matrix} & \begin{matrix} 1x \\ 1y \end{matrix} \\ \begin{matrix} K_{11}^{21} & K_{12}^{21} \\ K_{21}^{21} & K_{22}^{21} \end{matrix} & \begin{matrix} K_{11}^{22} & K_{12}^{22} \\ K_{21}^{22} & K_{22}^{22} \end{matrix} & \begin{matrix} K_{11}^{23} & K_{12}^{23} \\ K_{21}^{23} & K_{22}^{23} \end{matrix} & \begin{matrix} 2x \\ 2y \end{matrix} \\ \begin{matrix} K_{11}^{31} & K_{12}^{31} \\ K_{21}^{31} & K_{22}^{31} \end{matrix} & \begin{matrix} K_{11}^{32} & K_{12}^{32} \\ K_{21}^{32} & K_{22}^{32} \end{matrix} & \begin{matrix} K_{11}^{33} & K_{12}^{33} \\ K_{21}^{33} & K_{22}^{33} \end{matrix} & \begin{matrix} 3x \\ 3y \end{matrix} \end{bmatrix}$$

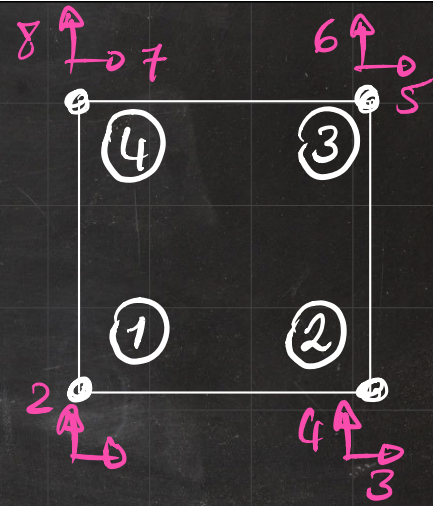
- 1 no NODE¹ X
- 2 no NODE¹ Y
- 3 no NODE² X
- 4 no NODE² Y
- 5 no NODE³ X
- 6 no NODE³ Y

D2Q4u4v



$$K = \begin{bmatrix} K_{11} & K_{12} & K_{13} & K_{14} \\ K_{21} & K_{22} & K_{23} & K_{24} \\ K_{31} & K_{32} & K_{33} & K_{34} \\ K_{41} & K_{42} & K_{43} & K_{44} \end{bmatrix}$$

8×8
Non-symmetric
 4×4 4×2



$$K_{ij} = \begin{bmatrix} K_{11}^{ij} & K_{12}^{ij} \\ K_{21}^{ij} & K_{22}^{ij} \end{bmatrix} = \begin{bmatrix} K_{xx}^{ij} & K_{xy}^{ij} \\ K_{yx}^{ij} & K_{yy}^{ij} \end{bmatrix}$$

1, 2 - NODE¹_{xy}
3, 4 - NODE²_{xy}
5, 6 - NODE³_{xy}
7, 8 - NODE⁴_{xy}

D2TR3N

$[K]_{6 \times 6}$

D2TR6N

$[K]_{12 \times 12}$

D2QU4N

$[K]_{8 \times 8}$

D2QU8N

$[K]_{16 \times 16}$

D2QU9N

$[K]_{18 \times 18}$

$$[K]_{ac}^{ij} = \int_B [N_{,ac}]_b^i E_{abcd} [N_{,ac}]_d^j dA$$

PROBLEMS TO ADDRESS

↳ INTEGRAL ↳ GAUSS QUADRATURE RULE

↳ $f(x)$ ↳ $x \rightarrow \xi$

↳ E_{abcd} ↳ ?

D2TR 3N

$[K]_{6 \times 6}$

D2TR 6N

$[K]_{12 \times 12}$

D2QU 4N

$[K]_{8 \times 8}$

D2QU 8N

$[K]_{8 \times 8}$

D2QU 9N

$[K]_{16 \times 16}$

$[K]_{18 \times 18}$

$[K]_{ac}^{ij} = \int_B [N_{,ac}]_b^i E_{abcd} [N_{,ac}]_d^j dA$

$E_{abcd} = \frac{E}{2(1+\nu)} [\delta_{ad} \delta_{bc} + \delta_{ac} \delta_{bd}]$

CONSTITUTIVE TENSOR

4th. O.

2x2x2x2 = 16 COMPONENTS

$+ \frac{E\nu}{1-\nu^2} \delta_{ab} \delta_{cd}$

Young's Modulus

ν : Poisson's Ratio

δ : Kronecker Delta

$$[K]_{ac}^{ij} = \int_B [N_{,ac}]_b^i E_{abcd} [N_{,ac}]_d^j dA$$

$$E_{abcd} = \frac{E}{2[1+\nu]} [\delta_{ad} \delta_{bc} + \delta_{ac} \delta_{bd}] + \frac{E\nu}{1-\nu^2} \delta_{ab} \delta_{cd}$$

$$E_{1111} = \frac{E}{[1-\nu^2]}$$

$$E_{1112} = 0$$

$$E_{1121} = 0$$

$$E_{1122} = \frac{E\nu}{[1-\nu^2]}$$

$$E_{1211} = 0$$

$$E_{1212} = \frac{E}{2[1+\nu]}$$

$$E_{1221} = \frac{E}{2[1+\nu]}$$

$$E_{1222} = 0$$

$$E_{2111} = 0$$

$$E_{2112} = \frac{E}{2[1+\nu]}$$

$$E_{2121} = \frac{E}{2[1+\nu]}$$

$$E_{2122} = 0$$

$$E_{2211} = \frac{E\nu}{[1-\nu^2]}$$

$$E_{2212} = 0$$

$$E_{2221} = 0$$

$$E_{2222} = \frac{E}{[1-\nu^2]}$$

$$[K]_{ac}^{ij} = \int_B [N_{,ac}^i]_b E_{abcd} [N_{,ac}^j]_d dA$$

$N_{,ac}^i$

$x = x(\xi)$

$x = x(\xi, \eta)$

$y = y(\xi, \eta)$

↳

$$\begin{bmatrix} \frac{\partial N^i}{\partial x} \\ \frac{\partial N^i}{\partial y} \end{bmatrix}$$

$$\frac{\partial N^i}{\partial x} = \frac{\partial N^i}{\partial \xi} \frac{\partial \xi}{\partial x} + \frac{\partial N^i}{\partial \eta} \frac{\partial \eta}{\partial x}$$

$$\frac{\partial N^i}{\partial y} = \frac{\partial N^i}{\partial \xi} \frac{\partial \xi}{\partial y} + \frac{\partial N^i}{\partial \eta} \frac{\partial \eta}{\partial y}$$

$\xi = \xi(x, y)$

$\eta = \eta(x, y)$

$\xi = \xi(x)$

$$[K]_{ac}^{ij} = \int_B [N_{,ac}^i]_b E_{abcd} [N_{,ac}^j]_d dA$$

$$[K]_{ac}^{ij} = \int_B [\mathbb{J} \cdot N_{,\xi}^i]_b E_{abcd} [\mathbb{J} \cdot N_{,\xi}^j]_d dA$$

$$\mathbb{J} = \frac{\partial x}{\partial \xi} \quad \hookrightarrow x = x(\xi) \quad \nearrow x = N^s \xi^s$$

$$\hookrightarrow N^s(\xi, \eta)$$

$$[K]_{ac}^{ij} = \int_B [\mathcal{J} \cdot N_{,a}^i]_b E_{abcd} [\mathcal{J} \cdot N_{,c}^j]_d dA$$

$$J_{11} = \frac{\partial x}{\partial \xi} = x^1 \frac{\partial N^1}{\partial \xi} + x^2 \frac{\partial N^2}{\partial \xi} + \dots + x^{NPE} \frac{\partial N^{NPE}}{\partial \xi}$$

$$J_{12} = \frac{\partial x}{\partial \eta} = x^1 \frac{\partial N^1}{\partial \eta} + x^2 \frac{\partial N^2}{\partial \eta} + \dots + x^{NPE} \frac{\partial N^{NPE}}{\partial \eta}$$

$$J_{21} = \frac{\partial y}{\partial \xi} = y^1 \frac{\partial N^1}{\partial \xi} + y^2 \frac{\partial N^2}{\partial \xi} + \dots + y^{NPE} \frac{\partial N^{NPE}}{\partial \xi}$$

$$J_{22} = \frac{\partial y}{\partial \eta} = y^1 \frac{\partial N^1}{\partial \eta} + y^2 \frac{\partial N^2}{\partial \eta} + \dots + y^{NPE} \frac{\partial N^{NPE}}{\partial \eta}$$

$$[K]_{ac}^{ij} = \int_B [\mathcal{J} \cdot N_{,a}^i]_b E_{abcd} [\mathcal{J} \cdot N_{,c}^j]_d dA$$

$$\underbrace{\begin{bmatrix} \mathcal{J}_{11} & \mathcal{J}_{21} \\ \mathcal{J}_{12} & \mathcal{J}_{22} \end{bmatrix}}_{\mathcal{J}^t} = \begin{bmatrix} \frac{\partial N^1}{\partial \xi} & \frac{\partial N^2}{\partial \xi} & \dots & \frac{\partial N^{NPE}}{\partial \xi} \\ \frac{\partial N^1}{\partial \eta} & \frac{\partial N^2}{\partial \eta} & \dots & \frac{\partial N^{NPE}}{\partial \eta} \end{bmatrix} \begin{bmatrix} \alpha^1 & y^1 \\ \alpha^2 & y^2 \\ \vdots & \vdots \\ \alpha^{NPE} & y^{NPE} \end{bmatrix}$$

$2 \times NPE$
 $NPE \times 2$

$$K_{ac}^{ij} = \sum_{gp=1}^{GPE} \left[\mathbb{J}^{-t} \cdot N_{,\xi}^i \right]_b E_{abcd} \left[\mathbb{J}^{-t} \cdot N_{,\xi}^j \right]_d \text{Det } \mathbb{J} \times \alpha_{gp} \times \frac{1}{2}$$

JACOBIAN $\frac{\partial x}{\partial \xi}$ \rightarrow $\mathbb{J} = \begin{bmatrix} x^1 \dots x^{NPE} \\ y^1 \dots y^{NPE} \end{bmatrix} \begin{bmatrix} N_{,\xi}^1 & N_{,\eta}^1 \\ \vdots & \vdots \\ N_{,\xi}^{NPE} & N_{,\eta}^{NPE} \end{bmatrix}$ IF TRIANGLE

$$E_{abcd} = \frac{E}{2[1+\nu]} [\delta_{ad} \delta_{bc} + \delta_{ac} \delta_{bd}] + \frac{E\nu}{1-\nu^2} \delta_{ab} \delta_{cd}$$

$$\left. \begin{array}{l} E_{1111} = \frac{E}{[1-\nu^2]} \\ E_{1211} = 0 \\ E_{2111} = 0 \\ E_{2211} = \frac{E\nu}{[1-\nu^2]} \end{array} \right\} \begin{array}{l} E_{1112} = 0 \\ E_{1212} = \frac{E}{2[1+\nu]} \\ E_{2112} = \frac{E}{2[1+\nu]} \\ E_{2212} = 0 \end{array} \begin{array}{l} E_{1121} = 0 \\ E_{1221} = \frac{E}{2[1+\nu]} \\ E_{2121} = \frac{E}{2[1+\nu]} \\ E_{2221} = 0 \end{array} \begin{array}{l} E_{1122} = \frac{E\nu}{[1-\nu^2]} \\ E_{1222} = 0 \\ E_{2122} = 0 \\ E_{2222} = \frac{E}{[1-\nu^2]} \end{array}$$

FINITE ELEMENT METHOD

FINITE ELEMENT METHOD

2D FEM

formulation summary
& understanding via examples

K_{ac}^{ij}

stiffness between
direction "a" of node "i" &
direction "c" of node "j"

$$\equiv \frac{\delta F_a^i}{\delta u_c^j}$$

Quadrilateral Elements

$$K_{ac}^{ij} = \sum_{gp=1}^{GPE} [\mathbf{J}^{-T} \cdot N_{,\xi}^i]_b E_{abcd} [\mathbf{J}^{-T} \cdot N_{,\xi}^j]_d \text{Det} \mathbf{J} \alpha_{gp}$$

Triangular Elements

$$K_{ac}^{ij} = \sum_{gp=1}^{GPE} [\mathbf{J}^{-T} \cdot N_{,\xi}^i]_b E_{abcd} [\mathbf{J}^{-T} \cdot N_{,\xi}^j]_d \text{Det} \mathbf{J} \alpha_{gp} \times \frac{1}{2}$$

Quadrilateral Elements

$$K_{ac}^{ij} = \sum_{gp=1}^{\text{GPE}} [\mathbf{J}^{-T} \cdot N_{,\xi}^i]_b E_{abcd} [\mathbf{J}^{-T} \cdot N_{,\xi}^j]_d \text{Det} \mathbf{J} \alpha_{gp}$$

Triangular Elements

$$K_{ac}^{ij} = \sum_{gp=1}^{\text{GPE}} [\mathbf{J}^{-T} \cdot N_{,\xi}^i]_b E_{abcd} [\mathbf{J}^{-T} \cdot N_{,\xi}^j]_d \text{Det} \mathbf{J} \alpha_{gp} \times \frac{1}{2}$$

$$E_{abcd} = \frac{E}{2[1 + \nu]} [\delta_{ad} \delta_{bc} + \delta_{ac} \delta_{bd}] + \frac{E \nu}{1 - \nu^2} \delta_{ab} \delta_{cd}$$

2D plane strain constitutive tensor components

$$E_{abcd} = \frac{E}{2[1 + \nu]} [\delta_{ad} \delta_{bc} + \delta_{ac} \delta_{bd}] + \frac{E \nu}{1 - \nu^2} \delta_{ab} \delta_{cd}$$

$$E_{1111} = \frac{E}{1 - \nu^2} \quad E_{1112} = 0 \quad E_{1121} = 0 \quad E_{1122} = \frac{E \nu}{1 - \nu^2}$$

$$E_{1211} = 0 \quad E_{1212} = \frac{E}{2[1 + \nu]} \quad E_{1221} = \frac{E}{2[1 + \nu]} \quad E_{1222} = 0$$

$$E_{2111} = 0 \quad E_{2112} = \frac{E}{2[1 + \nu]} \quad E_{2121} = \frac{E}{2[1 + \nu]} \quad E_{2122} = 0$$

$$E_{2211} = \frac{E \nu}{1 - \nu^2} \quad E_{2212} = 0 \quad E_{2221} = 0 \quad E_{2222} = \frac{E}{1 - \nu^2}$$

K_{ac}^{ij} stiffness between
direction "a" of node "i" &
direction "c" of node "j"

Quadrilateral Elements

$$K_{ac}^{ij} = \sum_{gp=1}^{GPE} [\mathbf{J}^{-T} \cdot N_{,\xi}^i]_b E_{abcd} [\mathbf{J}^{-T} \cdot N_{,\xi}^j]_d \text{Det} \mathbf{J} \alpha_{gp}$$

Triangular Elements

$$K_{ac}^{ij} = \sum_{gp=1}^{GPE} [\mathbf{J}^{-T} \cdot N_{,\xi}^i]_b E_{abcd} [\mathbf{J}^{-T} \cdot N_{,\xi}^j]_d \text{Det} \mathbf{J} \alpha_{gp} \times \frac{1}{2}$$

$$\mathbf{J} = \frac{\partial \mathbf{x}}{\partial \xi} = \begin{bmatrix} x^1 & x^2 & \dots & x^{\text{NPE}} \\ y^1 & y^2 & \dots & y^{\text{NPE}} \end{bmatrix} \begin{bmatrix} N_{\xi}^1 & N_{\eta}^1 \\ N_{\xi}^2 & N_{\eta}^2 \\ \vdots & \vdots \\ N_{\xi}^{\text{NPE}} & N_{\eta}^{\text{NPE}} \end{bmatrix}$$

$$E_{abcd} = \frac{E}{2[1+\nu]} [\delta_{ad} \delta_{bc} + \delta_{ac} \delta_{bd}]$$

plane strain $+ \frac{E\nu}{1-\nu^2} \delta_{ab} \delta_{cd}$

$$K_{ac}^{ij}$$

stiffness between
direction "a" of node "i" &
direction "c" of node "j"

Quadrilateral
Elements

$$K_{ac}^{ij} = \sum_{gp=1}^{GPE} [\mathbf{J}^{-T} \cdot N_{,\xi}^i]_b E_{abcd} [\mathbf{J}^{-T} \cdot N_{,\xi}^j]_d \text{Det} \mathbf{J} \alpha_{gp}$$

Triangular
Elements

$$K_{ac}^{ij} = \sum_{gp=1}^{GPE} [\mathbf{J}^{-T} \cdot N_{,\xi}^i]_b E_{abcd} [\mathbf{J}^{-T} \cdot N_{,\xi}^j]_d \text{Det} \mathbf{J} \alpha_{gp} \times \frac{1}{2}$$

plane strain

$$E_{abcd} = \frac{E}{2[1+\nu]} [\delta_{ad} \delta_{bc} + \delta_{ac} \delta_{bd}] + \frac{E\nu}{1-\nu^2} \delta_{ab} \delta_{cd}$$


$$\mathbf{J} = \begin{bmatrix} x^1 & x^2 & \dots & x^{\text{NPE}} \\ y^1 & y^2 & \dots & y^{\text{NPE}} \end{bmatrix} \begin{bmatrix} N_{\xi}^1 & N_{\eta}^1 \\ N_{\xi}^2 & N_{\eta}^2 \\ \vdots & \vdots \\ N_{\xi}^{\text{NPE}} & N_{\eta}^{\text{NPE}} \end{bmatrix}$$

$$\begin{aligned} E_{1111} &= \frac{E}{1-\nu^2} & E_{1112} &= 0 & E_{1121} &= 0 & E_{1122} &= \frac{E\nu}{1-\nu^2} \\ E_{1211} &= 0 & E_{1212} &= \frac{E}{2[1+\nu]} & E_{1221} &= \frac{E}{2[1+\nu]} & E_{1222} &= 0 \\ E_{2111} &= 0 & E_{2112} &= \frac{E}{2[1+\nu]} & E_{2121} &= \frac{E}{2[1+\nu]} & E_{2122} &= 0 \\ E_{2211} &= \frac{E\nu}{1-\nu^2} & E_{2212} &= 0 & E_{2221} &= 0 & E_{2222} &= \frac{E}{1-\nu^2} \end{aligned}$$

Understanding Jacobian

$$\mathbf{J} = \frac{\partial \mathbf{x}}{\partial \boldsymbol{\xi}} = \begin{bmatrix} x^1 & x^2 & \dots & x^{\text{NPE}} \\ y^1 & y^2 & \dots & y^{\text{NPE}} \end{bmatrix} \begin{bmatrix} N_{\xi}^1 & N_{\eta}^1 \\ N_{\xi}^2 & N_{\eta}^2 \\ \vdots & \vdots \\ N_{\xi}^{\text{NPE}} & N_{\eta}^{\text{NPE}} \end{bmatrix}$$

linear mapping
from natural space
to physical space



$$d\mathbf{x} = \mathbf{J} d\boldsymbol{\xi} \quad J = \text{Det} \mathbf{J} = \frac{dA_{\mathbf{x}}}{dA_{\boldsymbol{\xi}}}$$

often, the (scalar) determinant
of the Jacobian matrix is also
referred to as Jacobian

...
the scalar Jacobian is a linear
mapping between the area
elements from the natural
space to the physical space

Example (D2TR3N) ... constant strain triangle ... stiffness

Triangular
Elements

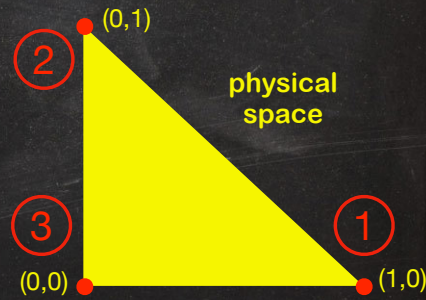
$$K_{ac}^{ij} = \sum_{gp=1}^{GPE} [J^{-T} \cdot N_{,\xi}^i]_b E_{abcd} [J^{-T} \cdot N_{,\xi}^j]_d \text{Det}J \alpha_{gp} \times \frac{1}{2}$$

$$E_{abcd} = \frac{E}{2[1+\nu]} [\delta_{ad}\delta_{bc} + \delta_{ac}\delta_{bd}] + \frac{E\nu}{1-\nu^2} \delta_{ab}\delta_{cd}$$

plane strain

$$K = \begin{bmatrix} K^{11} & K^{12} & K^{13} \\ K^{21} & K^{22} & K^{23} \\ K^{31} & K^{32} & K^{33} \end{bmatrix} = \begin{bmatrix} K_{11}^{11} & K_{12}^{11} & K_{11}^{12} & K_{12}^{12} & K_{11}^{13} & K_{12}^{13} \\ K_{21}^{11} & K_{22}^{11} & K_{21}^{12} & K_{22}^{12} & K_{21}^{13} & K_{22}^{13} \\ K_{11}^{21} & K_{12}^{21} & K_{11}^{22} & K_{12}^{22} & K_{11}^{23} & K_{12}^{23} \\ K_{21}^{21} & K_{22}^{21} & K_{21}^{22} & K_{22}^{22} & K_{21}^{23} & K_{22}^{23} \\ K_{11}^{31} & K_{12}^{31} & K_{11}^{32} & K_{12}^{32} & K_{11}^{33} & K_{12}^{33} \\ K_{21}^{31} & K_{22}^{31} & K_{21}^{32} & K_{22}^{32} & K_{21}^{33} & K_{22}^{33} \end{bmatrix}$$

$$J = \begin{bmatrix} x^1 & x^2 & \dots & x^{NPE} \\ y^1 & y^2 & \dots & y^{NPE} \end{bmatrix} \begin{bmatrix} N_{\xi}^1 & N_{\eta}^1 \\ N_{\xi}^2 & N_{\eta}^2 \\ \vdots & \vdots \\ N_{\xi}^{NPE} & N_{\eta}^{NPE} \end{bmatrix}$$



Example (D2TR3N) ... constant strain triangle ... stiffness

Triangular
Elements

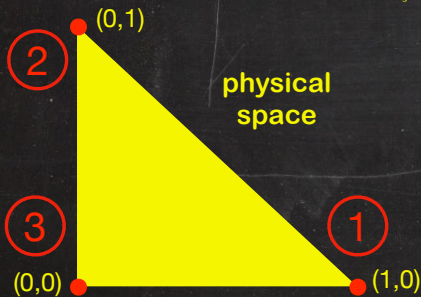
$$K_{ac}^{ij} = \sum_{gp=1}^{GPE} [\mathbf{J}^{-T} \cdot N_{,\xi}^i]_b E_{abcd} [\mathbf{J}^{-T} \cdot N_{,\xi}^j]_d \text{Det} \mathbf{J} \alpha_{gp} \times \frac{1}{2}$$

$$E_{abcd} = \frac{E}{2[1+\nu]} [\delta_{ad} \delta_{bc} + \delta_{ac} \delta_{bd}] + \frac{E\nu}{1-\nu^2} \delta_{ab} \delta_{cd}$$

plane strain

$$\mathbf{J} = \begin{bmatrix} x^1 & x^2 & \dots & x^{NPE} \\ y^1 & y^2 & \dots & y^{NPE} \end{bmatrix} \begin{bmatrix} N_{\xi}^1 & N_{\eta}^1 \\ N_{\xi}^2 & N_{\eta}^2 \\ \vdots & \vdots \\ N_{\xi}^{NPE} & N_{\eta}^{NPE} \end{bmatrix}$$

$E_{1111} = \frac{E}{1-\nu^2}$	$E_{1112} = 0$	$E_{1121} = 0$	$E_{1122} = \frac{E\nu}{1-\nu^2}$
$E_{1211} = 0$	$E_{1212} = \frac{E}{2[1+\nu]}$	$E_{1221} = \frac{E}{2[1+\nu]}$	$E_{1222} = 0$
$E_{2111} = 0$	$E_{2112} = \frac{E}{2[1+\nu]}$	$E_{2121} = \frac{E}{2[1+\nu]}$	$E_{2122} = 0$
$E_{2211} = \frac{E\nu}{1-\nu^2}$	$E_{2212} = 0$	$E_{2221} = 0$	$E_{2222} = \frac{E}{1-\nu^2}$



$N^1 = \xi$	natural space	$N_{,\xi}^1 = 1$	$N_{,\eta}^1 = 0$
$N^2 = \eta$		$N_{,\xi}^2 = 0$	$N_{,\eta}^2 = 1$
$N^3 = (1 - \xi - \eta)$		$N_{,\xi}^3 = -1$	$N_{,\eta}^3(\xi, \eta) = -1$

$$\mathbf{K} = \begin{bmatrix} K^{11} & K^{12} & K^{13} \\ K^{21} & K^{22} & K^{23} \\ K^{31} & K^{32} & K^{33} \end{bmatrix}$$

Example (D2TR3N) ... linear triangular element

Triangular
Elements

$$K_{ac}^{ij} = \sum_{gp=1}^{GPE} [\mathbf{J}^{-T} \cdot N_{,\xi}^i]_b E_{abcd} [\mathbf{J}^{-T} \cdot N_{,\xi}^j]_d \text{Det}\mathbf{J} \alpha_{gp} \times \frac{1}{2}$$

$$\mathbf{K} = \begin{bmatrix} \frac{E}{2[1-\nu^2]} & 0 & 0 & \frac{E\nu}{2[1-\nu^2]} & -\frac{E}{2[1-\nu^2]} & -\frac{E\nu}{2[1-\nu^2]} \\ 0 & \frac{E}{4[\nu+1]} & \frac{E}{4[\nu+1]} & 0 & -\frac{E}{4[\nu+1]} & -\frac{E}{4[\nu+1]} \\ 0 & \frac{E}{4[\nu+1]} & \frac{E}{4[\nu+1]} & 0 & -\frac{E}{4[\nu+1]} & -\frac{E}{4[\nu+1]} \\ \frac{E\nu}{2[1-\nu^2]} & 0 & 0 & \frac{E}{2[1-\nu^2]} & -\frac{E\nu}{2[1-\nu^2]} & -\frac{E}{2[1-\nu^2]} \\ -\frac{E}{2[1-\nu^2]} & -\frac{E}{4[\nu+1]} & -\frac{E}{4[\nu+1]} & -\frac{E\nu}{2[1-\nu^2]} & \frac{E[3-\nu]}{4[1-\nu^2]} & \frac{E}{4[1-\nu]} \\ -\frac{E\nu}{2[1-\nu^2]} & -\frac{E}{4[\nu+1]} & -\frac{E}{4[\nu+1]} & -\frac{E}{2[1-\nu^2]} & \frac{E}{4[1-\nu]} & \frac{E[3-\nu]}{4[1-\nu^2]} \end{bmatrix}$$

$$E_{abcd} = \frac{E}{2[1+\nu]} [\delta_{ad}\delta_{bc} + \delta_{ac}\delta_{bd}] + \frac{E\nu}{1-\nu^2} \delta_{ab}\delta_{cd}$$

plane strain

$$\mathbf{J} = \begin{bmatrix} x^1 & x^2 & \dots & x^{\text{NPE}} \\ y^1 & y^2 & \dots & y^{\text{NPE}} \end{bmatrix} \begin{bmatrix} N_{\xi}^1 & N_{\eta}^1 \\ N_{\xi}^2 & N_{\eta}^2 \\ \vdots & \vdots \\ N_{\xi}^{\text{NPE}} & N_{\eta}^{\text{NPE}} \end{bmatrix}$$

$$\mathbf{x}^1 = [1, 0]$$

$$\mathbf{x}^2 = [0, 1]$$

$$\mathbf{x}^3 = [0, 0]$$

Example (D2TR3N) ... linear triangular element

... using one Gauss point ...

Triangular
Elements

$$K_{ac}^{ij} = \sum_{gp=1}^{GPE} [\mathbf{J}^{-T} \cdot N_{,\xi}^i]_b E_{abcd} [\mathbf{J}^{-T} \cdot N_{,\xi}^j]_d \text{Det}\mathbf{J} \alpha_{gp} \times \frac{1}{2}$$

$$E_{abcd} = \frac{E}{2[1+\nu]} [\delta_{ad}\delta_{bc} + \delta_{ac}\delta_{bd}] + \frac{E\nu}{1-\nu^2} \delta_{ab}\delta_{cd}$$

plane strain

$$\mathbf{K} = \begin{bmatrix} 5.000 & 0.000 & 0.000 & 0.000 & -5.000 & 0.000 \\ 0.000 & 2.500 & 2.500 & 0.000 & -2.500 & -2.500 \\ 0.000 & 2.500 & 2.500 & 0.000 & -2.500 & -2.500 \\ 0.000 & 0.000 & 0.000 & 5.000 & 0.000 & -5.000 \\ -5.000 & -2.500 & -2.500 & 0.000 & 7.500 & 2.500 \\ 0.000 & -2.500 & -2.500 & -5.000 & 2.500 & 7.500 \end{bmatrix}$$

$$\mathbf{J} = \begin{bmatrix} x^1 & x^2 & \dots & x^{NPE} \\ y^1 & y^2 & \dots & y^{NPE} \end{bmatrix} \begin{bmatrix} N_{\xi}^1 & N_{\eta}^1 \\ N_{\xi}^2 & N_{\eta}^2 \\ \vdots & \vdots \\ N_{\xi}^{NPE} & N_{\eta}^{NPE} \end{bmatrix}$$

$$\mathbf{x}^1 = [1, 0]$$

$$\mathbf{x}^2 = [0, 1]$$

$$\mathbf{x}^3 = [0, 0]$$

$$E = 10, \nu = 0$$

Example (D2TR3N) ... linear triangular element

... using one Gauss point ...

Triangular
Elements

$$K_{ac}^{ij} = \sum_{gp=1}^{GPE} [\mathbf{J}^{-T} \cdot \mathbf{N}_{,\xi}^i]_b E_{abcd} [\mathbf{J}^{-T} \cdot \mathbf{N}_{,\xi}^j]_d \text{Det}\mathbf{J} \alpha_{gp} \times \frac{1}{2}$$

$$E_{abcd} = \frac{E}{2[1+\nu]} [\delta_{ad}\delta_{bc} + \delta_{ac}\delta_{bd}] + \frac{E\nu}{1-\nu^2} \delta_{ab}\delta_{cd}$$

plane strain

$$\mathbf{K} = \begin{bmatrix} 6.667 & 0.000 & 0.000 & 3.333 & -6.667 & -3.333 \\ 0.000 & 1.667 & 1.667 & 0.000 & -1.667 & -1.667 \\ 0.000 & 1.667 & 1.667 & 0.000 & -1.667 & -1.667 \\ 3.333 & 0.000 & 0.000 & 6.667 & -3.333 & -6.667 \\ -6.667 & -1.667 & -1.667 & -3.333 & 8.333 & 5.000 \\ -3.333 & -1.667 & -1.667 & -6.667 & 5.000 & 8.333 \end{bmatrix}$$

$$\mathbf{J} = \begin{bmatrix} x^1 & x^2 & \dots & x^{NPE} \\ y^1 & y^2 & \dots & y^{NPE} \end{bmatrix} \begin{bmatrix} N_{\xi}^1 & N_{\eta}^1 \\ N_{\xi}^2 & N_{\eta}^2 \\ \vdots & \vdots \\ N_{\xi}^{NPE} & N_{\eta}^{NPE} \end{bmatrix}$$

$$\mathbf{x}^1 = [1, 0]$$

$$\mathbf{x}^2 = [0, 1]$$

$$\mathbf{x}^3 = [0, 0]$$

$$E = 10, \nu = 0.5$$

Example (D2QU4N) ... bilinear quadrilateral element

Quadrilateral Elements

$$K_{ac}^{ij} = \sum_{gp=1}^{GPE} [J^{-T} \cdot N_{,\xi}^i]_b E_{abcd} [J^{-T} \cdot N_{,\xi}^j]_d \text{Det} J \alpha_{gp}$$

$$E_{abcd} = \frac{E}{2[1+\nu]} [\delta_{ad} \delta_{bc} + \delta_{ac} \delta_{bd}] + \frac{E\nu}{1-\nu^2} \delta_{ab} \delta_{cd}$$

plane strain

$$K = \begin{bmatrix} K^{11} & K^{12} & K^{13} & K^{14} \\ K^{21} & K^{22} & K^{23} & K^{24} \\ K^{31} & K^{32} & K^{33} & K^{34} \\ K^{41} & K^{42} & K^{43} & K^{44} \end{bmatrix} = \begin{bmatrix} K_{11}^{11} & K_{12}^{11} & K_{11}^{12} & K_{12}^{12} & K_{11}^{13} & K_{12}^{13} & K_{11}^{14} & K_{12}^{14} \\ K_{21}^{11} & K_{22}^{11} & K_{21}^{12} & K_{22}^{12} & K_{21}^{13} & K_{22}^{13} & K_{21}^{14} & K_{22}^{14} \\ K_{11}^{21} & K_{12}^{21} & K_{11}^{22} & K_{12}^{22} & K_{11}^{23} & K_{12}^{23} & K_{11}^{24} & K_{12}^{24} \\ K_{21}^{21} & K_{22}^{21} & K_{21}^{22} & K_{22}^{22} & K_{21}^{23} & K_{22}^{23} & K_{21}^{24} & K_{22}^{24} \\ K_{11}^{31} & K_{12}^{31} & K_{11}^{32} & K_{12}^{32} & K_{11}^{33} & K_{12}^{33} & K_{11}^{34} & K_{12}^{34} \\ K_{21}^{31} & K_{22}^{31} & K_{21}^{32} & K_{22}^{32} & K_{21}^{33} & K_{22}^{33} & K_{21}^{34} & K_{22}^{34} \\ K_{11}^{41} & K_{12}^{41} & K_{11}^{42} & K_{12}^{42} & K_{11}^{43} & K_{12}^{43} & K_{11}^{44} & K_{12}^{44} \\ K_{21}^{41} & K_{22}^{41} & K_{21}^{42} & K_{22}^{42} & K_{21}^{43} & K_{22}^{43} & K_{21}^{44} & K_{22}^{44} \end{bmatrix}$$

$$J = \begin{bmatrix} x^1 & x^2 & \dots & x^{NPE} \\ y^1 & y^2 & \dots & y^{NPE} \end{bmatrix} \begin{bmatrix} N_{\xi}^1 & N_{\eta}^1 \\ N_{\xi}^2 & N_{\eta}^2 \\ \vdots & \vdots \\ N_{\xi}^{NPE} & N_{\eta}^{NPE} \end{bmatrix}$$

! ATTENTION !
 element stiffness matrix is symmetric and its determinant is zero

Example (D2QU4N) ... bilinear quadrilateral element

... using one Gauss point ...

Quadrilateral
Elements

$$K_{ac}^{ij} = \sum_{gp=1}^{GPE} [J^{-T} \cdot N_{,\xi}^i]_b E_{abcd} [J^{-T} \cdot N_{,\xi}^j]_d \text{Det} J \alpha_{gp}$$

$$E_{abcd} = \frac{E}{2[1+\nu]} [\delta_{ad} \delta_{bc} + \delta_{ac} \delta_{bd}]$$

plane strain

$$+ \frac{E\nu}{1-\nu^2} \delta_{ab} \delta_{cd}$$

$$K = \begin{bmatrix} \frac{E[\nu-3]}{8[\nu^2-1]} & -\frac{E}{8[\nu-1]} & \frac{E}{8[\nu-1]} & \frac{E[1-3\nu]}{8[\nu^2-1]} & -\frac{E[\nu-3]}{8[\nu^2-1]} & \frac{E}{8[\nu-1]} & -\frac{E}{8[\nu-1]} & -\frac{E[1-3\nu]}{8[\nu^2-1]} \\ -\frac{E}{8[\nu-1]} & \frac{E[\nu-3]}{8[\nu^2-1]} & -\frac{E[1-3\nu]}{8[\nu^2-1]} & \frac{E}{8[\nu-1]} & \frac{E}{8[\nu-1]} & -\frac{E[\nu-3]}{8[\nu^2-1]} & \frac{E[1-3\nu]}{8[\nu^2-1]} & \frac{E}{8[\nu-1]} \\ \frac{E}{8[\nu-1]} & -\frac{E[1-3\nu]}{8[\nu^2-1]} & \frac{E[\nu-3]}{8[\nu^2-1]} & \frac{E}{8[\nu-1]} & -\frac{E}{8[\nu-1]} & \frac{E[1-3\nu]}{8[\nu^2-1]} & -\frac{E[\nu-3]}{8[\nu^2-1]} & -\frac{E}{8[\nu-1]} \\ \frac{E[1-3\nu]}{8[\nu^2-1]} & \frac{E}{8[\nu-1]} & \frac{E}{8[\nu-1]} & \frac{E[\nu-3]}{8[\nu^2-1]} & -\frac{E[1-3\nu]}{8[\nu^2-1]} & \frac{E}{8[\nu-1]} & -\frac{E}{8[\nu-1]} & -\frac{E[\nu-3]}{8[\nu^2-1]} \\ \frac{E[\nu-3]}{8[\nu^2-1]} & \frac{E}{8[\nu-1]} & -\frac{E}{8[\nu-1]} & \frac{E[1-3\nu]}{8[\nu^2-1]} & \frac{E[\nu-3]}{8[\nu^2-1]} & -\frac{E}{8[\nu-1]} & \frac{E}{8[\nu-1]} & \frac{E[1-3\nu]}{8[\nu^2-1]} \\ \frac{E}{8[\nu-1]} & -\frac{E[\nu-3]}{8[\nu^2-1]} & \frac{E[1-3\nu]}{8[\nu^2-1]} & \frac{E}{8[\nu-1]} & -\frac{E}{8[\nu-1]} & \frac{E[\nu-3]}{8[\nu^2-1]} & -\frac{E[1-3\nu]}{8[\nu^2-1]} & -\frac{E}{8[\nu-1]} \\ -\frac{E}{8[\nu-1]} & \frac{E[1-3\nu]}{8[\nu^2-1]} & -\frac{E[\nu-3]}{8[\nu^2-1]} & -\frac{E}{8[\nu-1]} & \frac{E}{8[\nu-1]} & -\frac{E[1-3\nu]}{8[\nu^2-1]} & \frac{E[\nu-3]}{8[\nu^2-1]} & \frac{E}{8[\nu-1]} \\ \frac{E[1-3\nu]}{8[\nu^2-1]} & \frac{E}{8[\nu-1]} & -\frac{E}{8[\nu-1]} & -\frac{E[\nu-3]}{8[\nu^2-1]} & \frac{E[1-3\nu]}{8[\nu^2-1]} & -\frac{E}{8[\nu-1]} & \frac{E}{8[\nu-1]} & \frac{E[\nu-3]}{8[\nu^2-1]} \end{bmatrix}$$

$$J = \begin{bmatrix} x^1 & x^2 & \dots & x^{NPE} \\ y^1 & y^2 & \dots & y^{NPE} \end{bmatrix} \begin{bmatrix} N_{\xi}^1 & N_{\eta}^1 \\ N_{\xi}^2 & N_{\eta}^2 \\ \vdots & \vdots \\ N_{\xi}^{NPE} & N_{\eta}^{NPE} \end{bmatrix}$$

$$\mathbf{x}^1 = [-1, -1]$$

$$\mathbf{x}^2 = [1, -1]$$

$$\mathbf{x}^3 = [1, 1]$$

$$\mathbf{x}^4 = [-1, 1]$$

Example (D2QU4N) ... bilinear quadrilateral element

... using one Gauss point ...

Quadrilateral
Elements

$$K_{ac}^{ij} = \sum_{gp=1}^{GPE} [J^{-T} \cdot N_{,\xi}^i]_b E_{abcd} [J^{-T} \cdot N_{,\xi}^j]_d \text{Det} J \alpha_{gp}$$

$$E_{abcd} = \frac{E}{2[1+\nu]} [\delta_{ad} \delta_{bc} + \delta_{ac} \delta_{bd}] + \frac{E\nu}{1-\nu^2} \delta_{ab} \delta_{cd}$$

plane strain

$$K = \begin{bmatrix} 3.750 & 1.250 & -1.250 & -1.250 & -3.750 & -1.250 & 1.250 & 1.250 \\ 1.250 & 3.750 & 1.250 & 1.250 & -1.250 & -3.750 & -1.250 & -1.250 \\ -1.250 & 1.250 & 3.750 & -1.250 & 1.250 & -1.250 & -3.750 & 1.250 \\ -1.250 & 1.250 & -1.250 & 3.750 & 1.250 & -1.250 & 1.250 & -3.750 \\ -3.750 & -1.250 & 1.250 & 1.250 & 3.750 & 1.250 & -1.250 & -1.250 \\ -1.250 & -3.750 & -1.250 & -1.250 & 1.250 & 3.750 & 1.250 & 1.250 \\ 1.250 & -1.250 & -3.750 & 1.250 & -1.250 & 1.250 & 3.750 & -1.250 \\ 1.250 & -1.250 & 1.250 & -3.750 & -1.250 & 1.250 & -1.250 & 3.750 \end{bmatrix}$$

$$J = \begin{bmatrix} x^1 & x^2 & \dots & x^{NPE} \\ y^1 & y^2 & \dots & y^{NPE} \end{bmatrix} \begin{bmatrix} N_{\xi}^1 & N_{\eta}^1 \\ N_{\xi}^2 & N_{\eta}^2 \\ \vdots & \vdots \\ N_{\xi}^{NPE} & N_{\eta}^{NPE} \end{bmatrix}$$

- $\mathbf{x}^1 = [-1, -1]$
- $\mathbf{x}^2 = [1, -1]$
- $\mathbf{x}^3 = [1, 1]$
- $\mathbf{x}^4 = [-1, 1]$

$$E = 10, \nu = 0$$

Example (D2QU4N) ... bilinear quadrilateral element

... using four Gauss points ...

Quadrilateral
Elements

$$K_{ac}^{ij} = \sum_{gp=1}^{GPE} [J^{-T} \cdot N_{,\xi}^i]_b E_{abcd} [J^{-T} \cdot N_{,\xi}^j]_d \text{Det} J \alpha_{gp}$$

$$E_{abcd} = \frac{E}{2[1+\nu]} [\delta_{ad} \delta_{bc} + \delta_{ac} \delta_{bd}]$$

plane strain

$$+ \frac{E\nu}{1-\nu^2} \delta_{ab} \delta_{cd}$$

$$K = \begin{bmatrix} 5.000 & 1.250 & -2.500 & -1.250 & -2.500 & -1.250 & 0.000 & 1.250 \\ 1.250 & 5.000 & 1.250 & -0.000 & -1.250 & -2.500 & -1.250 & -2.500 \\ -2.500 & 1.250 & 5.000 & -1.250 & 0.000 & -1.250 & -2.500 & 1.250 \\ -1.250 & -0.000 & -1.250 & 5.000 & 1.250 & -2.500 & 1.250 & -2.500 \\ -2.500 & -1.250 & 0.000 & 1.250 & 5.000 & 1.250 & -2.500 & -1.250 \\ -1.250 & -2.500 & -1.250 & -2.500 & 1.250 & 5.000 & 1.250 & -0.000 \\ 0.000 & -1.250 & -2.500 & 1.250 & -2.500 & 1.250 & 5.000 & -1.250 \\ 1.250 & -2.500 & 1.250 & -2.500 & -1.250 & -0.000 & -1.250 & 5.000 \end{bmatrix}$$

$$J = \begin{bmatrix} x^1 & x^2 & \dots & x^{NPE} \\ y^1 & y^2 & \dots & y^{NPE} \end{bmatrix} \begin{bmatrix} N_{\xi}^1 & N_{\eta}^1 \\ N_{\xi}^2 & N_{\eta}^2 \\ \vdots & \vdots \\ N_{\xi}^{NPE} & N_{\eta}^{NPE} \end{bmatrix}$$

$$\mathbf{x}^1 = [-1, -1]$$

$$\mathbf{x}^2 = [1, -1]$$

$$\mathbf{x}^3 = [1, 1]$$

$$\mathbf{x}^4 = [-1, 1]$$

$$E = 10, \nu = 0$$

Example (D2QU4N) ... bilinear quadrilateral element

... using nine Gauss points ...

Quadrilateral
Elements

$$K_{ac}^{ij} = \sum_{gp=1}^{GPE} [J^{-T} \cdot N_{,\xi}^i]_b E_{abcd} [J^{-T} \cdot N_{,\xi}^j]_d \text{Det} J \alpha_{gp}$$

$$E_{abcd} = \frac{E}{2[1+\nu]} [\delta_{ad} \delta_{bc} + \delta_{ac} \delta_{bd}] + \frac{E\nu}{1-\nu^2} \delta_{ab} \delta_{cd}$$

plane strain

$$K = \begin{bmatrix} 5.000 & 1.250 & -2.500 & -1.250 & -2.500 & -1.250 & -0.000 & 1.250 \\ 1.250 & 5.000 & 1.250 & -0.000 & -1.250 & -2.500 & -1.250 & -2.500 \\ -2.500 & 1.250 & 5.000 & -1.250 & -0.000 & -1.250 & -2.500 & 1.250 \\ -1.250 & -0.000 & -1.250 & 5.000 & 1.250 & -2.500 & 1.250 & -2.500 \\ -2.500 & -1.250 & -0.000 & 1.250 & 5.000 & 1.250 & -2.500 & -1.250 \\ -1.250 & -2.500 & -1.250 & -2.500 & 1.250 & 5.000 & 1.250 & -0.000 \\ -0.000 & -1.250 & -2.500 & 1.250 & -2.500 & 1.250 & 5.000 & -1.250 \\ 1.250 & -2.500 & 1.250 & -2.500 & -1.250 & -0.000 & -1.250 & 5.000 \end{bmatrix}$$

$$J = \begin{bmatrix} x^1 & x^2 & \dots & x^{NPE} \\ y^1 & y^2 & \dots & y^{NPE} \end{bmatrix} \begin{bmatrix} N_{\xi}^1 & N_{\eta}^1 \\ N_{\xi}^2 & N_{\eta}^2 \\ \vdots & \vdots \\ N_{\xi}^{NPE} & N_{\eta}^{NPE} \end{bmatrix}$$

- $\mathbf{x}^1 = [-1, -1]$
- $\mathbf{x}^2 = [1, -1]$
- $\mathbf{x}^3 = [1, 1]$
- $\mathbf{x}^4 = [-1, 1]$

$$E = 10, \nu = 0$$

Example (D2QU4N) ... bilinear quadrilateral element

... using one Gauss point ...

Quadrilateral
Elements

$$K_{ac}^{ij} = \sum_{gp=1}^{GPE} [J^{-T} \cdot N_{,\xi}^i]_b E_{abcd} [J^{-T} \cdot N_{,\xi}^j]_d \text{Det} J \alpha_{gp}$$

$$E_{abcd} = \frac{E}{2[1+\nu]} [\delta_{ad} \delta_{bc} + \delta_{ac} \delta_{bd}] + \frac{E\nu}{1-\nu^2} \delta_{ab} \delta_{cd}$$

plane strain

$$K = \begin{bmatrix} 4.167 & 2.500 & -2.500 & 0.833 & -4.167 & -2.500 & 2.500 & -0.833 \\ 2.500 & 4.167 & -0.833 & 2.500 & -2.500 & -4.167 & 0.833 & -2.500 \\ -2.500 & -0.833 & 4.167 & -2.500 & 2.500 & 0.833 & -4.167 & 2.500 \\ 0.833 & 2.500 & -2.500 & 4.167 & -0.833 & -2.500 & 2.500 & -4.167 \\ -4.167 & -2.500 & 2.500 & -0.833 & 4.167 & 2.500 & -2.500 & 0.833 \\ -2.500 & -4.167 & 0.833 & -2.500 & 2.500 & 4.167 & -0.833 & 2.500 \\ 2.500 & 0.833 & -4.167 & 2.500 & -2.500 & -0.833 & 4.167 & -2.500 \\ -0.833 & -2.500 & 2.500 & -4.167 & 0.833 & 2.500 & -2.500 & 4.167 \end{bmatrix}$$

$$J = \begin{bmatrix} x^1 & x^2 & \dots & x^{NPE} \\ y^1 & y^2 & \dots & y^{NPE} \end{bmatrix} \begin{bmatrix} N_{\xi}^1 & N_{\eta}^1 \\ N_{\xi}^2 & N_{\eta}^2 \\ \vdots & \vdots \\ N_{\xi}^{NPE} & N_{\eta}^{NPE} \end{bmatrix}$$

$$\begin{aligned} \mathbf{x}^1 &= [-1, -1] \\ \mathbf{x}^2 &= [1, -1] \\ \mathbf{x}^3 &= [1, 1] \\ \mathbf{x}^4 &= [-1, 1] \end{aligned}$$

$$E = 10, \nu = 0.5$$

Example (D2QU4N) ... bilinear quadrilateral element

... using four Gauss points ...

Quadrilateral
Elements

$$K_{ac}^{ij} = \sum_{gp=1}^{GPE} [J^{-T} \cdot N_{,\xi}^i]_b E_{abcd} [J^{-T} \cdot N_{,\xi}^j]_d \text{Det} J \alpha_{gp}$$

$$E_{abcd} = \frac{E}{2[1+\nu]} [\delta_{ad} \delta_{bc} + \delta_{ac} \delta_{bd}] + \frac{E\nu}{1-\nu^2} \delta_{ab} \delta_{cd}$$

plane strain

$$K = \begin{bmatrix} 5.556 & 2.500 & -3.889 & 0.833 & -2.778 & -2.500 & 1.111 & -0.833 \\ 2.500 & 5.556 & -0.833 & 1.111 & -2.500 & -2.778 & 0.833 & -3.889 \\ -3.889 & -0.833 & 5.556 & -2.500 & 1.111 & 0.833 & -2.778 & 2.500 \\ 0.833 & 1.111 & -2.500 & 5.556 & -0.833 & -3.889 & 2.500 & -2.778 \\ -2.778 & -2.500 & 1.111 & -0.833 & 5.556 & 2.500 & -3.889 & 0.833 \\ -2.500 & -2.778 & 0.833 & -3.889 & 2.500 & 5.556 & -0.833 & 1.111 \\ 1.111 & 0.833 & -2.778 & 2.500 & -3.889 & -0.833 & 5.556 & -2.500 \\ -0.833 & -3.889 & 2.500 & -2.778 & 0.833 & 1.111 & -2.500 & 5.556 \end{bmatrix}$$

$$J = \begin{bmatrix} x^1 & x^2 & \dots & x^{NPE} \\ y^1 & y^2 & \dots & y^{NPE} \end{bmatrix} \begin{bmatrix} N_{\xi}^1 & N_{\eta}^1 \\ N_{\xi}^2 & N_{\eta}^2 \\ \vdots & \vdots \\ N_{\xi}^{NPE} & N_{\eta}^{NPE} \end{bmatrix}$$

$$\mathbf{x}^1 = [-1, -1]$$

$$\mathbf{x}^2 = [1, -1]$$

$$\mathbf{x}^3 = [1, 1]$$

$$\mathbf{x}^4 = [-1, 1]$$

$$E = 10, \nu = 0.5$$

Example (D2QU4N) ... bilinear quadrilateral element

... using nine Gauss points ...

Quadrilateral
Elements

$$K_{ac}^{ij} = \sum_{gp=1}^{GPE} [J^{-T} \cdot N_{,\xi}^i]_b E_{abcd} [J^{-T} \cdot N_{,\xi}^j]_d \text{Det} J \alpha_{gp}$$

$$E_{abcd} = \frac{E}{2[1+\nu]} [\delta_{ad} \delta_{bc} + \delta_{ac} \delta_{bd}] + \frac{E\nu}{1-\nu^2} \delta_{ab} \delta_{cd}$$

plane strain

$$K = \begin{bmatrix} 5.556 & 2.500 & -3.889 & 0.833 & -2.778 & -2.500 & 1.111 & -0.833 \\ 2.500 & 5.556 & -0.833 & 1.111 & -2.500 & -2.778 & 0.833 & -3.889 \\ -3.889 & -0.833 & 5.556 & -2.500 & 1.111 & 0.833 & -2.778 & 2.500 \\ 0.833 & 1.111 & -2.500 & 5.556 & -0.833 & -3.889 & 2.500 & -2.778 \\ -2.778 & -2.500 & 1.111 & -0.833 & 5.556 & 2.500 & -3.889 & 0.833 \\ -2.500 & -2.778 & 0.833 & -3.889 & 2.500 & 5.556 & -0.833 & 1.111 \\ 1.111 & 0.833 & -2.778 & 2.500 & -3.889 & -0.833 & 5.556 & -2.500 \\ -0.833 & -3.889 & 2.500 & -2.778 & 0.833 & 1.111 & -2.500 & 5.556 \end{bmatrix}$$

$$J = \begin{bmatrix} x^1 & x^2 & \dots & x^{NPE} \\ y^1 & y^2 & \dots & y^{NPE} \end{bmatrix} \begin{bmatrix} N_{\xi}^1 & N_{\eta}^1 \\ N_{\xi}^2 & N_{\eta}^2 \\ \vdots & \vdots \\ N_{\xi}^{NPE} & N_{\eta}^{NPE} \end{bmatrix}$$

- $\mathbf{x}^1 = [-1, -1]$
- $\mathbf{x}^2 = [1, -1]$
- $\mathbf{x}^3 = [1, 1]$
- $\mathbf{x}^4 = [-1, 1]$

$$E = 10, \nu = 0.5$$

finite elements

Examples

Ali Javili

department of mechanical engineering

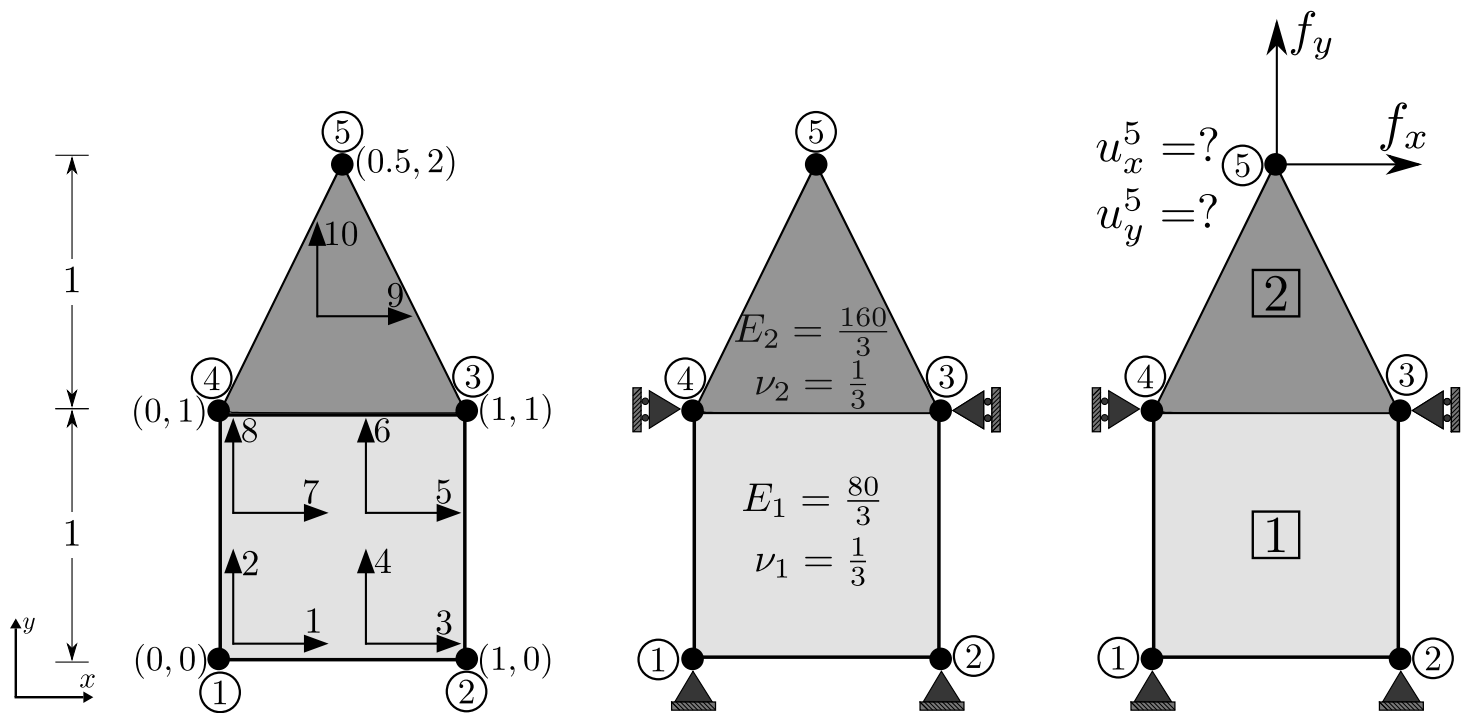
Bilkent university



Example 1

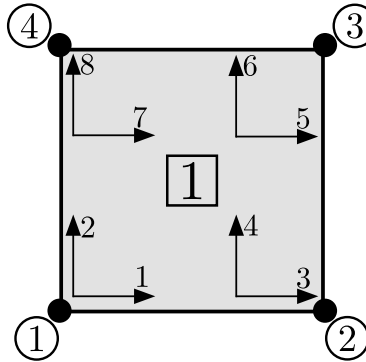
Consider the given structure consisting of a square and a triangle element. The element 1 is bilinear quadrilateral, also referred to as D2QU4N, and element 2 is linear triangular, also referred to as D2TR3N. The material parameters of each element are shown in the figure. The material properties for the square element and the triangle element are $E_1 = 80/3$, $\nu_1 = 1/3$ and $E_2 = 160/3$, $\nu_2 = 1/3$, respectively. The numbering of the nodes, elements and the global degrees are given in the figure. The nodes are numbered with a circle around them and elements are distinguished by square around them. **You must use exactly the same numbering for the degrees and elements given in the figure.** Homogeneous Dirichlet boundary conditions are prescribed on nodes 1 and 2 in both directions. Nodes 3 and 4 are fixed only in horizontal direction, but can freely move in vertical direction. The forces f_x and f_y are applied on node 5 in the horizontal and vertical directions, respectively.

Example 1

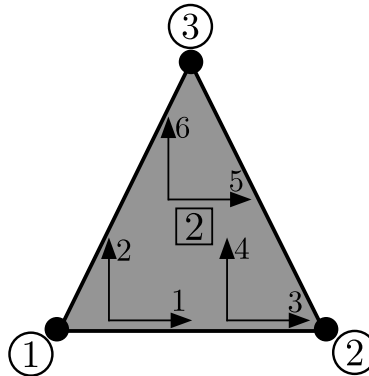


- Compute the 8×8 stiffness of the element 1 using one Gauss Point quadrature rule.
- Compute the 6×6 stiffness of the element 2 using one Gauss Point quadrature rule.
- Assemble the **reduced** stiffness for the entire system.
- Compute the displacement of node 5 in x and y directions for the three cases:
 - (1) $f_x = 1$, $f_y = 0$,
 - (2) $f_x = 0$, $f_y = 1$,
 - (3) $f_x = 1$, $f_y = 1$.

Example 1



$$\mathbb{K}_1 = \begin{bmatrix} 1x & 1y & 2x & 2y & 3x & 3y & 4x & 4y \\ 10 & 5 & -5 & 0 & -10 & -5 & 5 & 0 \\ 5 & 10 & 0 & 5 & -5 & -10 & 0 & -5 \\ -5 & 0 & 10 & -5 & 5 & 0 & -10 & 5 \\ 0 & 5 & -5 & 10 & 0 & -5 & 5 & -10 \\ -10 & -5 & 5 & 0 & 10 & 5 & -5 & 0 \\ -5 & -10 & 0 & -5 & 5 & 10 & 0 & 5 \\ 5 & 0 & -10 & 5 & -5 & 0 & 10 & -5 \\ 0 & -5 & 5 & -10 & 0 & 5 & -5 & 10 \end{bmatrix} \begin{matrix} 1x \\ 1y \\ 2x \\ 2y \\ 3x \\ 3y \\ 4x \\ 4y \end{matrix}$$



$$\mathbb{K}_2 = \begin{bmatrix}
 4x & 4y & 5x & 5y & 3x & 3y & \\
 32.05 & 10 & -27.5 & 0 & -5 & -10 & 4x \\
 10 & 17.5 & 0 & -2.5 & -10 & -15 & 4y \\
 -27.5 & 0 & 32.5 & -10 & -5 & 10 & 5x \\
 0 & -2.5 & -10 & 17.5 & 10 & -15 & 5y \\
 -5 & -10 & -5 & 10 & 10 & 0 & 3x \\
 -10 & -15 & 10 & -15 & 0 & 30 & 3y
 \end{bmatrix}$$

$$\mathbb{K}_{\text{tot}} = \begin{bmatrix} 10 & 5 & -5 & 0 & -10 & -5 & 5 & 0 & 0 & 0 \\ 5 & 10 & 0 & 5 & -5 & -10 & 0 & -5 & 0 & 0 \\ -5 & 0 & 10 & -5 & 5 & 0 & -10 & 5 & 0 & 0 \\ 0 & 5 & -5 & 10 & 0 & -5 & 5 & -10 & 0 & 0 \\ -10 & -5 & 5 & 0 & 42.5 & -5 & -32.5 & 0 & -5 & 10 \\ -5 & -10 & 0 & -5 & -5 & 27.5 & 0 & 2.5 & 10 & -15 \\ 5 & 0 & -10 & 5 & -23.5 & 0 & 42.5 & 5 & -5 & -10 \\ 0 & -5 & 5 & -10 & 0 & 2.5 & 5 & 27.5 & -10 & -15 \\ 0 & 0 & 0 & 0 & -5 & 10 & -5 & -10 & 10 & 0 \\ 0 & 0 & 0 & 0 & 10 & -15 & -10 & -15 & 0 & 30 \end{bmatrix}$$

$$\mathbb{K}_{\text{red}} = \begin{array}{cccc|c} & 4y & 3y & 5x & 5y & \\ \hline & 27.5 & 2.5 & -10 & -15 & 4y \\ & 2.5 & 27.5 & 10 & -15 & 3y \\ & -10 & 10 & 10 & 0 & 5x \\ & -15 & -15 & 0 & 30 & 5y \end{array}$$

$$\begin{bmatrix} 27.5 & 2.5 & -10 & -15 \\ 2.5 & 27.5 & 10 & -15 \\ -10 & 10 & 10 & 0 \\ -15 & -15 & 0 & 30 \end{bmatrix} \begin{bmatrix} u_y^4 \\ u_y^3 \\ u_x^5 \\ u_y^5 \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ f_x \\ f_y \end{bmatrix}$$

$$\left\{ \begin{array}{l} 27.5u_y^4 + 2.5u_y^3 - 10u_x^5 - 15u_y^5 = 0 \\ 2.5u_y^4 + 27.5u_y^3 + 10u_x^5 - 15u_y^5 = 0 \\ -10u_y^4 + 10u_y^3 + 10u_x^5 = f_x \\ -15u_y^4 - 15u_y^3 + 30u_y^5 = f_y \end{array} \right.$$

$f_x = 1, f_y = 0$	$f_x = 0, f_y = 1$	$f_x = 1, f_y = 1$
$u_y^4 = 0.2$	$u_y^4 = \frac{0.1}{3}$	$u_y^4 = \frac{0.7}{3}$
$u_y^3 = -0.2$	$u_y^3 = \frac{0.1}{3}$	$u_y^3 = -\frac{0.5}{3}$
$u_x^5 = 0.5$	$u_x^5 = 0$	$u_x^5 = \frac{1.5}{3}$
$u_y^5 = 0$	$u_y^5 = \frac{0.2}{3}$	$u_y^5 = \frac{0.2}{3}$