

# MECHANICS AND MATERIALS I

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22

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## Combined Loading

Section ... 8.2

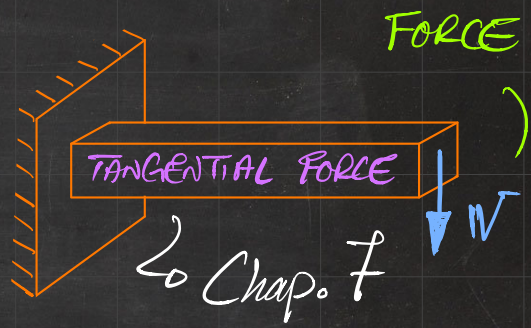
Chap. 8

[ Hibbeler 9th edition ]

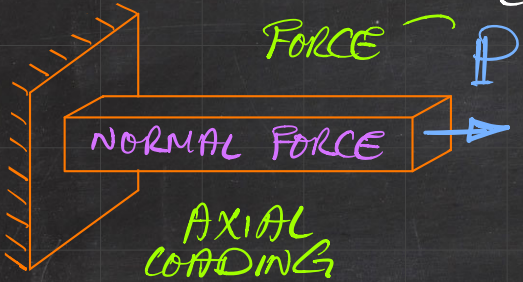
# COMBINED LOADING

Chap. 8

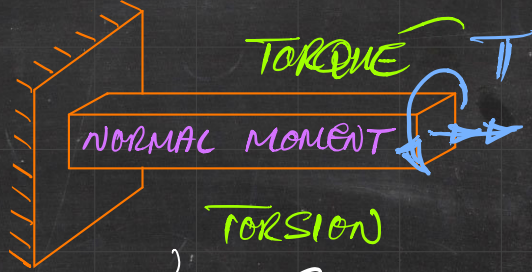
$$\tau = \frac{VQ}{It}$$



$$\delta = \frac{PL}{EA}$$



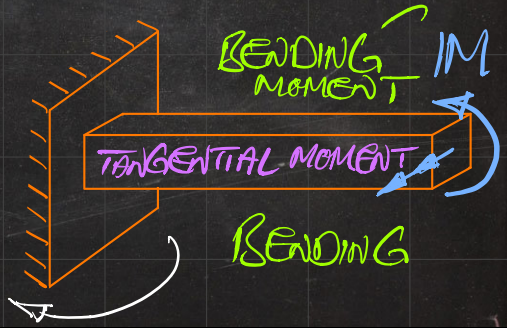
Chap. 4



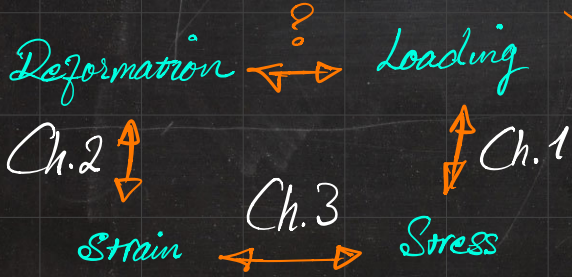
$$\phi = \frac{TL}{GJ}$$

$$\theta = \frac{ML}{EI}$$

Chap. 5



Chap. 6

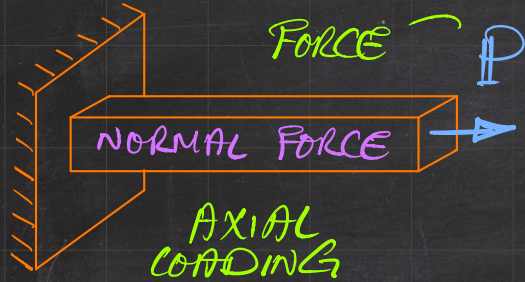




# COMBINED LOADING



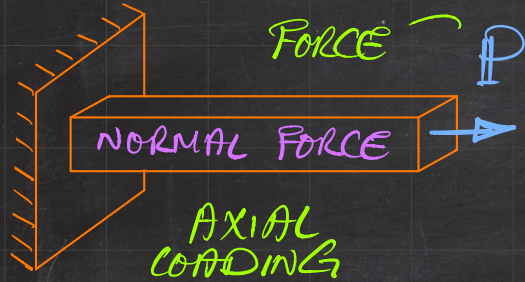
# COMBINED LOADING



$$\sigma = \frac{P}{A}$$

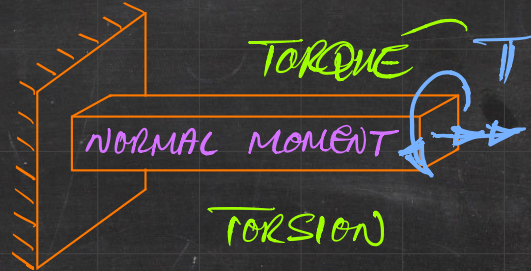
$$\delta = \frac{PL}{EA}$$

# COMBINED LOADING



$$\epsilon = \frac{P}{A}$$

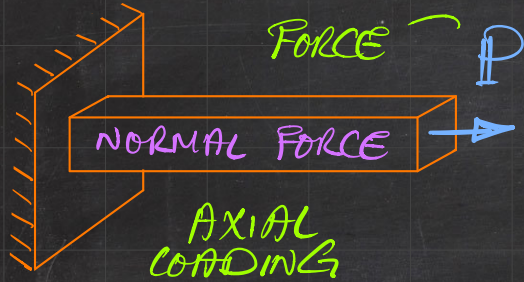
$$\delta = \frac{PL}{EA}$$



$$\tau = \frac{Tr}{J}$$

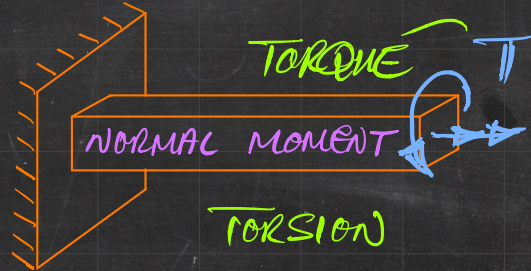
$$\phi = \frac{TL}{GJ}$$

# COMBINED LOADING



$$\epsilon = \frac{P}{A}$$

$$\delta = \frac{PL}{EA}$$

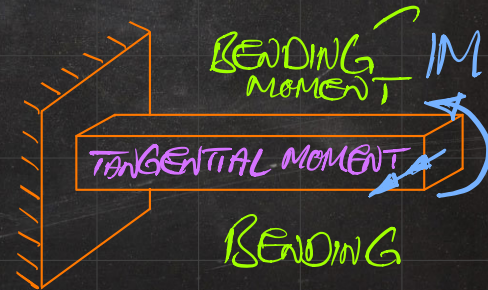


$$\tau = \frac{Tr}{J}$$

$$\phi = \frac{TL}{GJ}$$

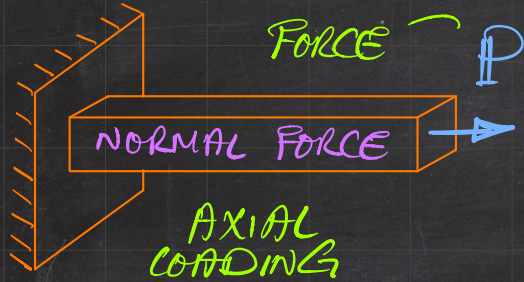
$$\epsilon = -\frac{My}{I}$$

$$\theta = \frac{ML}{EI}$$





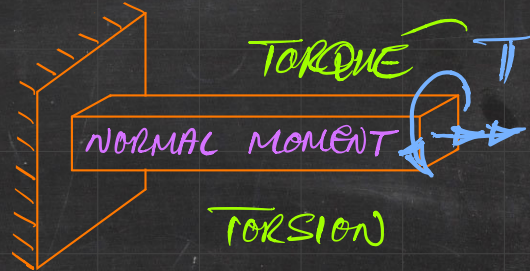
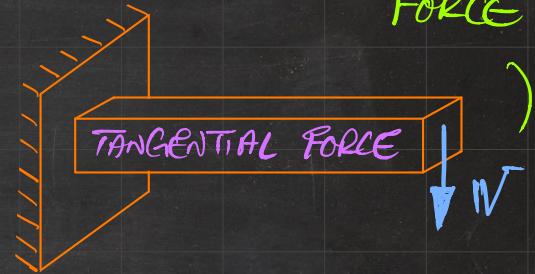
# COMBINED LOADING



$$\epsilon = \frac{P}{A}$$

$$\delta = \frac{PL}{EA}$$

$$\tau = \frac{VQ}{It}$$

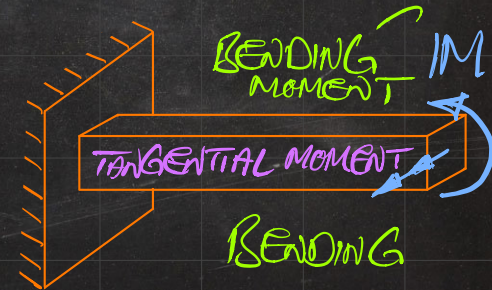


$$\tau = \frac{Tr}{J}$$

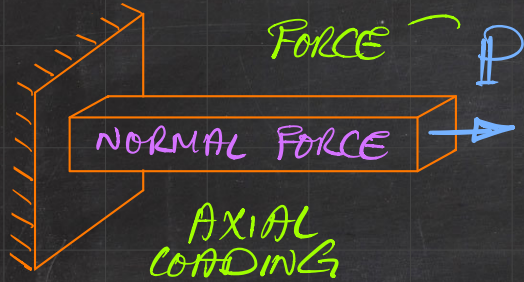
$$\phi = \frac{TL}{GJ}$$

$$\epsilon = -\frac{My}{I}$$

$$\theta = \frac{ML}{EI}$$



# COMBINED LOADING

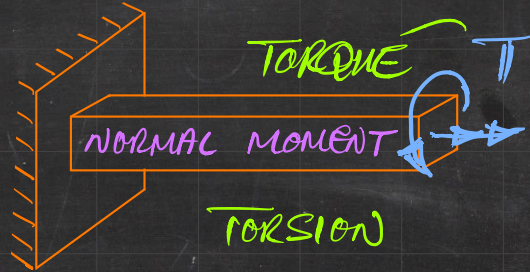
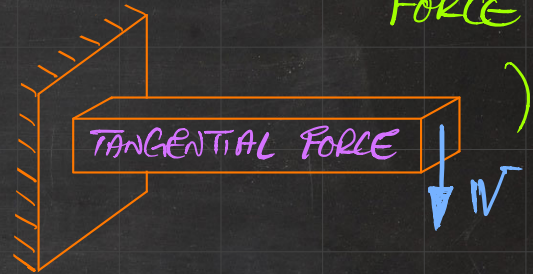


$$\epsilon = \frac{P}{A}$$

$$\delta = \frac{PL}{EA}$$

*in practice, a combination of all of these occurs!*

$$\tau = \frac{VQ}{It}$$

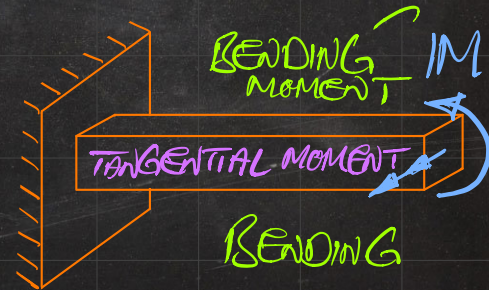


$$\tau = \frac{Tr}{J}$$

$$\phi = \frac{TL}{GJ}$$

$$\epsilon = -\frac{My}{I}$$

$$\theta = \frac{ML}{EI}$$



# COMBINED LOADING

\* AXIAL LOADING

$$\epsilon = \frac{P}{A}$$

$$\delta = \frac{PL}{EA}$$

\* TORSION

$$\tau = \frac{T r}{J}$$

$$\phi = \frac{TL}{GJ}$$

\* BENDING

$$\epsilon = -\frac{My}{I}$$

$$\theta = \frac{ML}{EI}$$

\* TRANSVERSE SHEAR

$$\tau = \frac{VQ}{It}$$



# COMBINED LOADING

\* AXIAL LOADING

$$\epsilon = \frac{P}{A}$$

$$\delta = \frac{PL}{EA}$$

\* TORSION

$$\tau = \frac{Tr}{J}$$

$$\phi = \frac{TL}{GJ}$$

\* BENDING

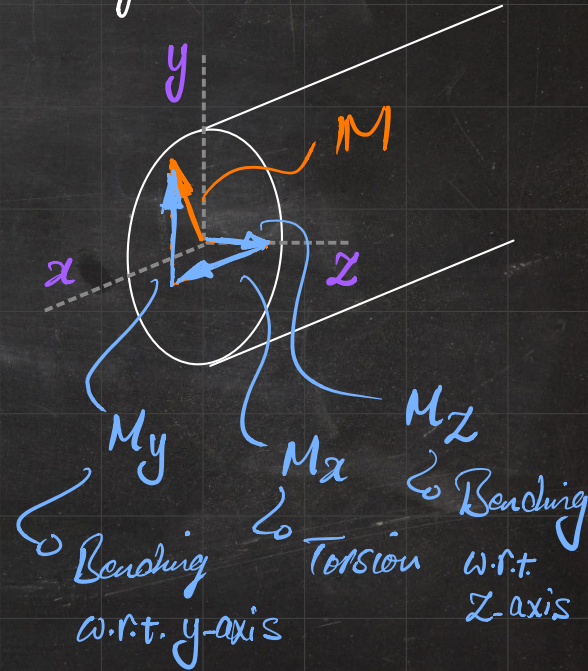
$$\epsilon = -\frac{My}{I}$$

$$\theta = \frac{ML}{EI}$$

\* TRANSVERSE SHEAR

$$\tau = \frac{VQ}{It}$$

e.g.



# COMBINED LOADING $\rightarrow$ STRESS-STATE ?

\* AXIAL LOADING

$$\epsilon = \frac{P}{A}$$

$$\delta = \frac{PL}{EA}$$

\* TORSION

$$\tau = \frac{Tr}{J}$$

$$\phi = \frac{TL}{GJ}$$

\* BENDING

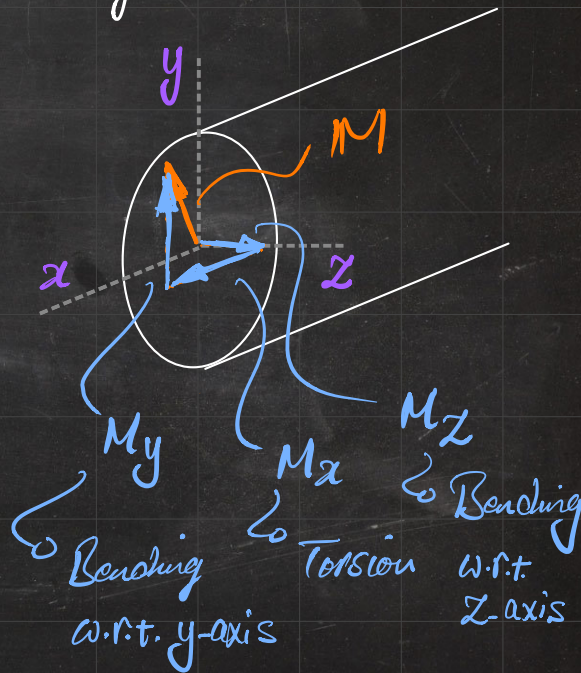
$$\epsilon = -\frac{My}{I}$$

$$\theta = \frac{ML}{EI}$$

\* TRANSVERSE SHEAR

$$\tau = \frac{VQ}{It}$$

e.g.



# COMBINED LOADING

\* AXIAL LOADING

$$\sigma = \frac{P}{A}$$

$$\delta = \frac{PL}{EA}$$

\* TORSION

$$\tau = \frac{Tr}{J}$$

$$\theta = \frac{TL}{GJ}$$

\* BENDING

$$\sigma = -\frac{My}{I}$$

$$\theta = \frac{ML}{EI}$$

\* TRANSVERSE SHEAR

$$\tau = \frac{VQ}{It}$$

TO COMPUTE  
STRESS-STATE

STEP 1)

FIND INTERNAL FORCES/MOMENTS

AT THE DESIRED CROSS-SECTION

STEP 2)

FIND STRESSES  $\sigma, \tau$  CAUSED

DUE TO INTERNAL REACTIONS

STEP 3)

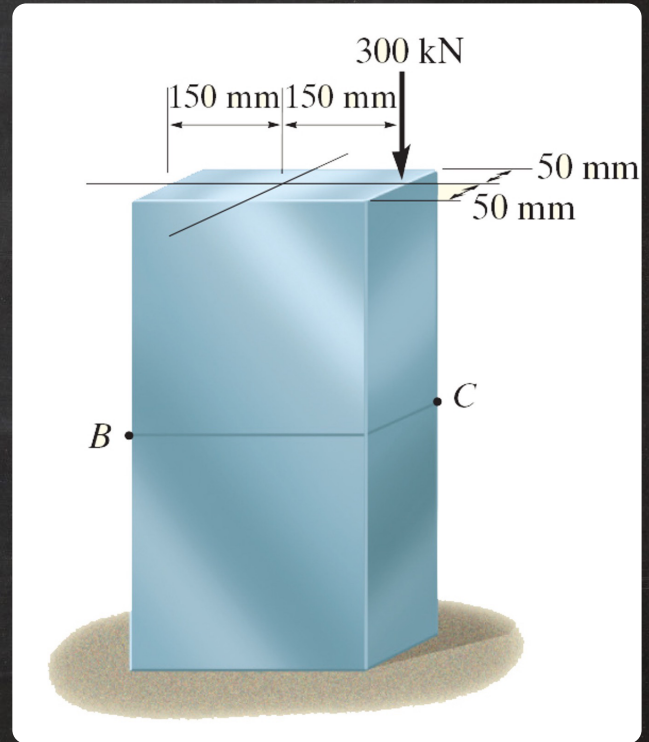
ADD UP THE EFFECTS

(Careful with signs!)



## Exercise 1 . [ similar to ... P. 418 ... 8.2 ]

FOR THE BLOCK SHOWN IN THE FIGURE,  
DETERMINE THE STATE OF STRESS AT  
POINTS B AND C.  
NEGLECT THE WEIGHT OF THE MEMBER.



STEP 1)

FIND INTERNAL FORCES/MOMENTS

AT THE DESIRED CROSS-SECTION

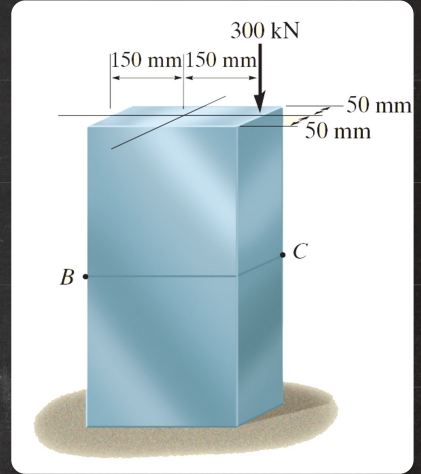
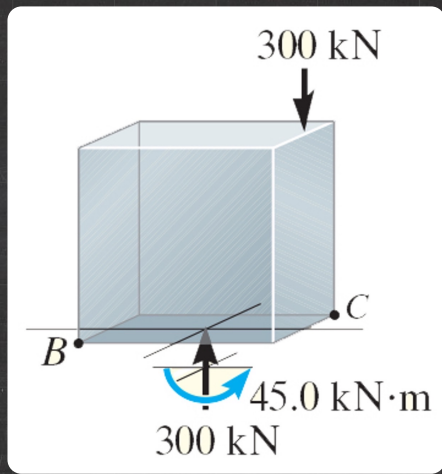
STEP 2)

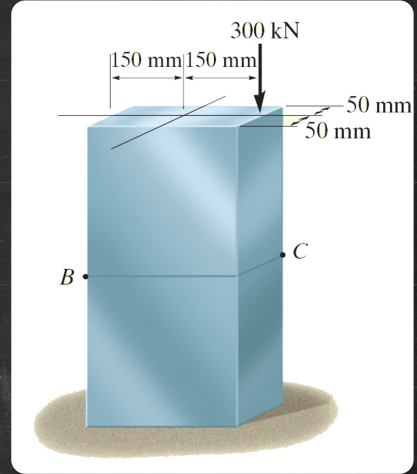
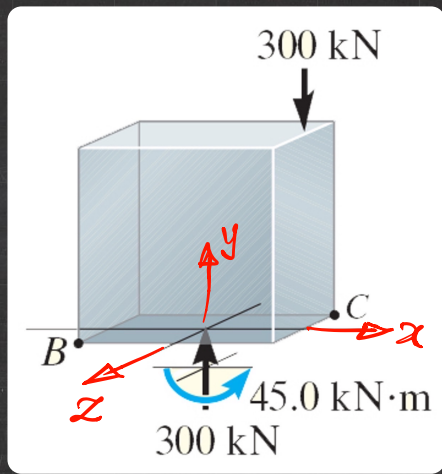
FIND STRESSES  $\sigma$ ,  $\tau$  CAUSED

DUE TO INTERNAL REACTIONS

STEP 3)

ADD UP THE EFFECTS



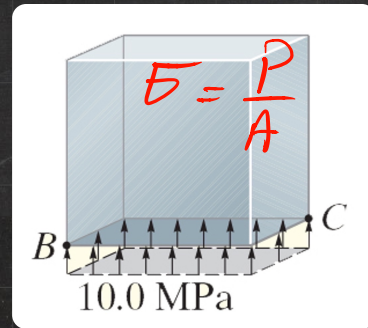
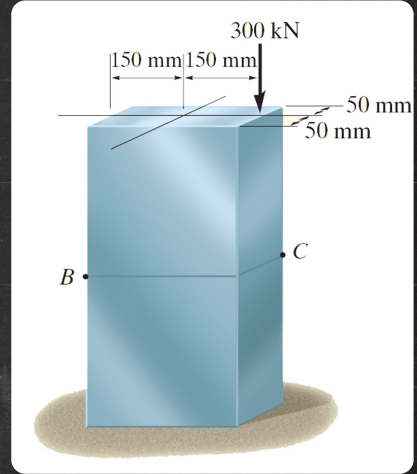
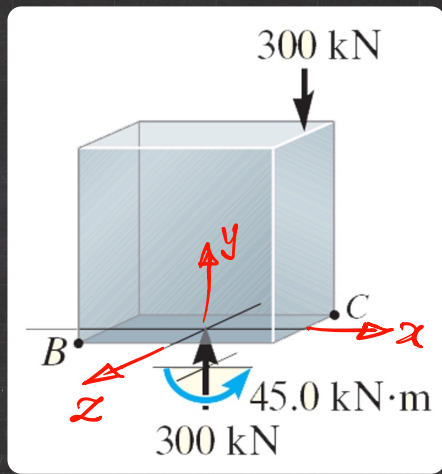




# AXIAL LOADING

$\sigma_B = \sigma_C = -10 \text{ MPa}$  *Compression*

$P = -300 \times 10^3, A = 0.03$



# AXIAL LOADING

Compression

$$\sigma_B = \sigma_C = -10 \text{ MPa}$$

$$P = -300 \times 10^3, A = 0.03$$

# BENDING

$$M_x = 45 \times 10^3 \quad (100 \text{ mm})$$

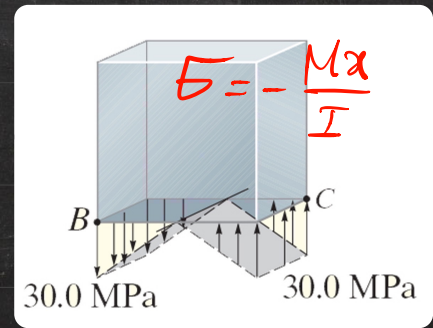
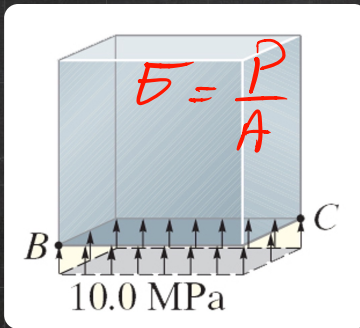
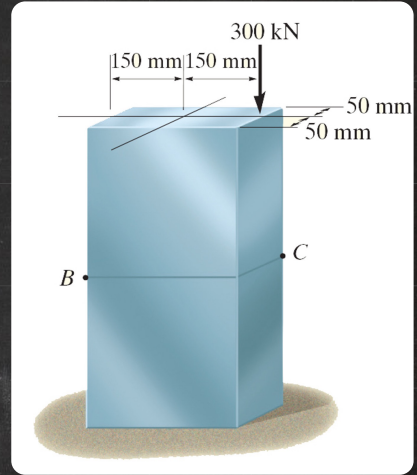
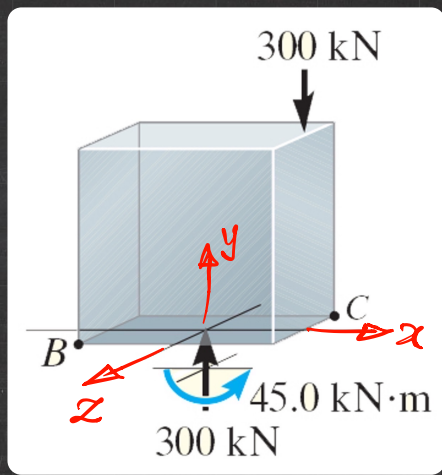
$$\sigma = -\frac{Mx}{I}$$

$$I_z = \frac{1}{12} bh^3 \quad (300 \text{ mm})$$

$$-200 x \text{ MPa}$$

$$225 \times 10^{-6} \text{ m}^4$$

$$\sigma_B = +30 \text{ MPa}, \sigma_C = -30 \text{ MPa}$$



# AXIAL LOADING

*Compression*

$$\sigma_B = \sigma_C = -10 \text{ MPa}$$

$$P = -300 \times 10^3, A = 0.03$$

# BENDING

$$M_z = 45 \times 10^3 \quad (100 \text{ mm})$$

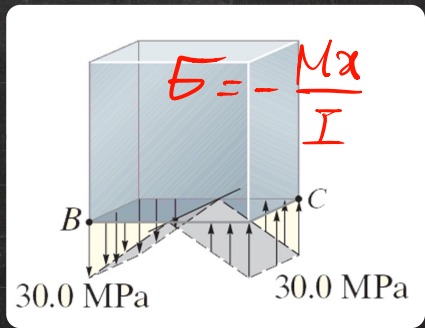
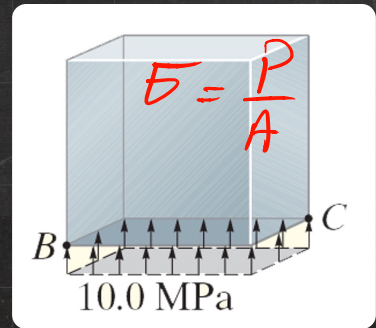
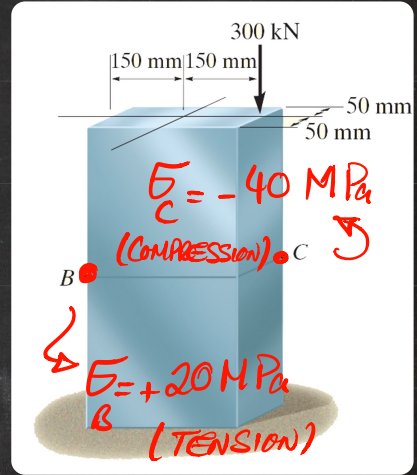
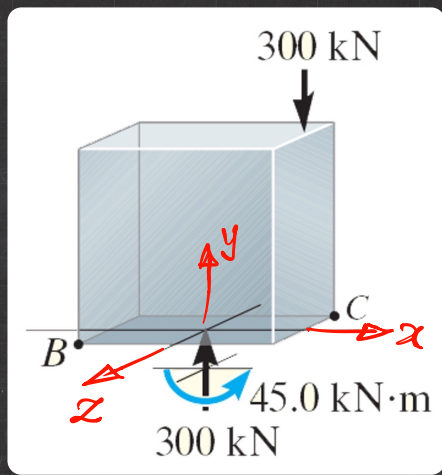
$$\sigma = -\frac{Mx}{I}$$

$$I_z = \frac{1}{12} bh^3 \quad (300 \text{ mm})$$

$$-200 x \quad \text{MPa}$$

$$225 \times 10^{-6} \text{ m}^4$$

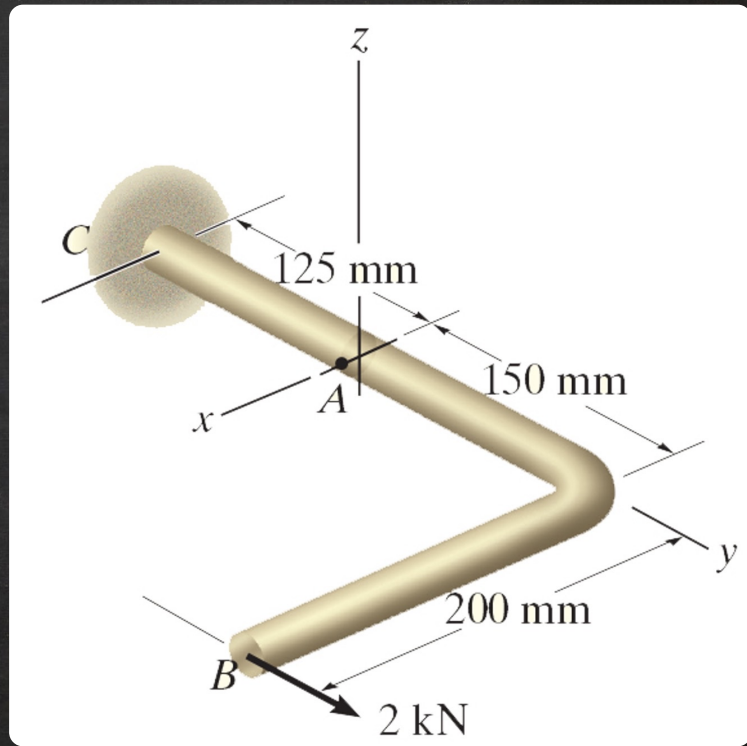
$$\sigma_B = +30 \text{ MPa}, \sigma_C = -30 \text{ MPa}$$





## Exercise 2 . [ similar to ... P. 422 ... 8.5 ]

DETERMINE THE STATE OF STRESS AT POINT A ON THE SOLID ROD SHOWN IN THE FIGURE. THE RADIUS OF THE SHAFT IS 20 mm.



STEP 1)

FIND INTERNAL FORCES/MOMENTS

AT THE DESIRED CROSS-SECTION

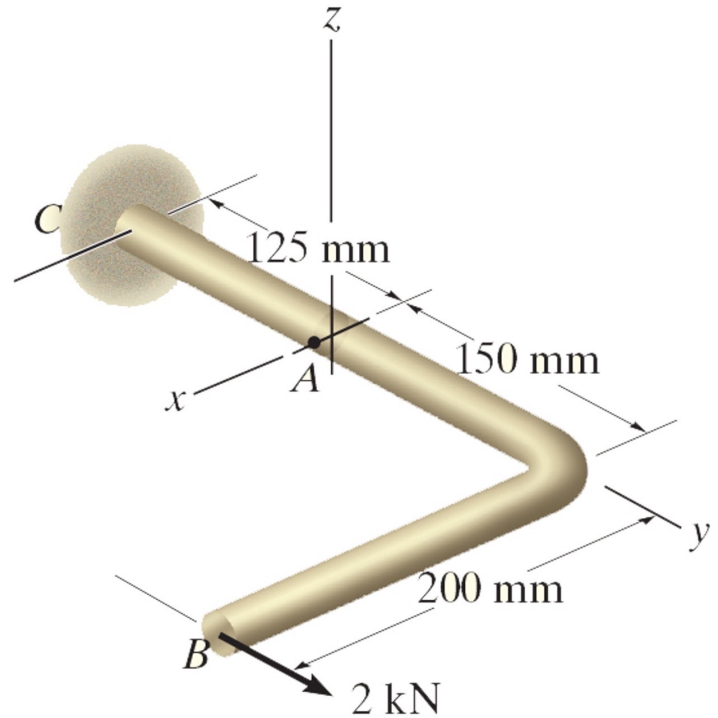
STEP 2)

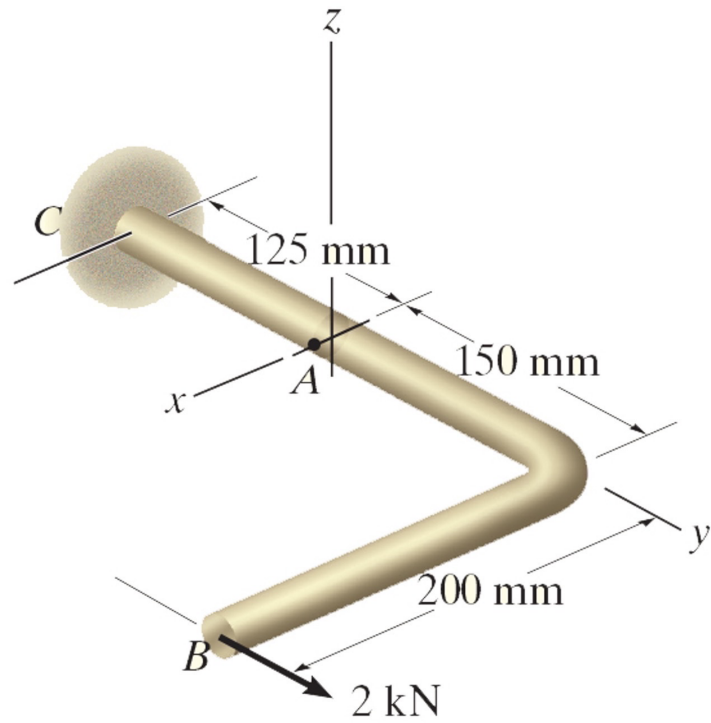
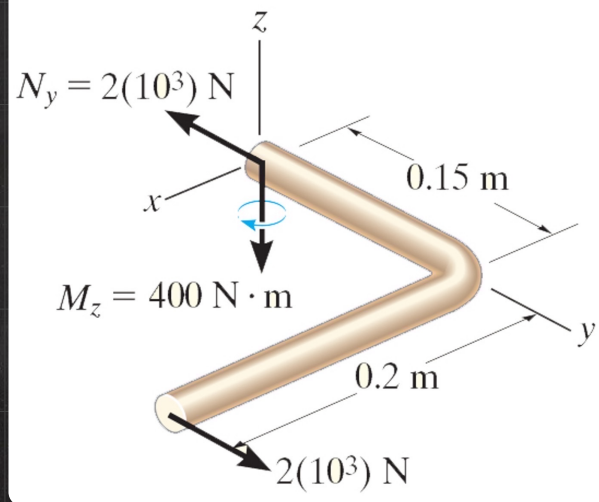
FIND STRESSES  $\sigma$ ,  $\tau$  CAUSED

DUE TO INTERNAL REACTIONS

STEP 3)

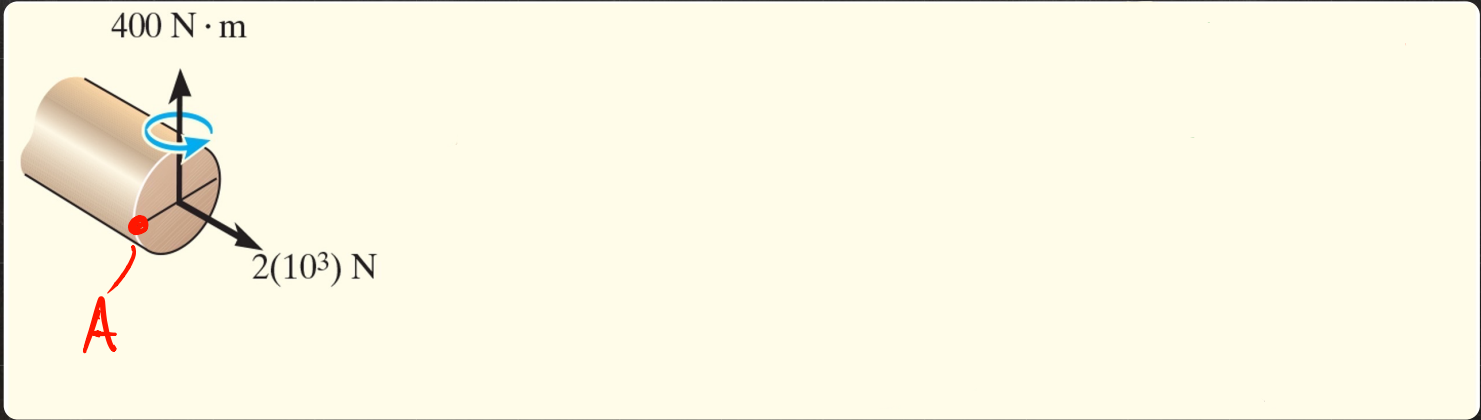
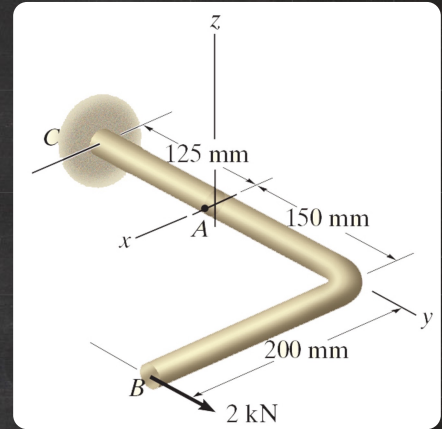
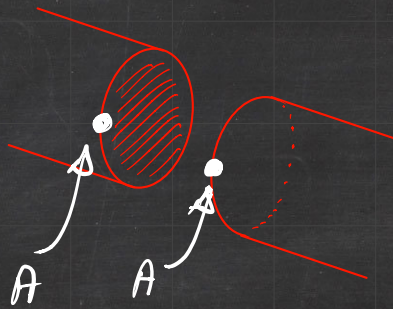
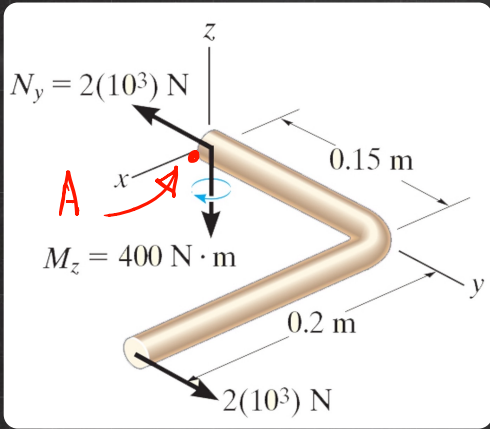
ADD UP THE EFFECTS

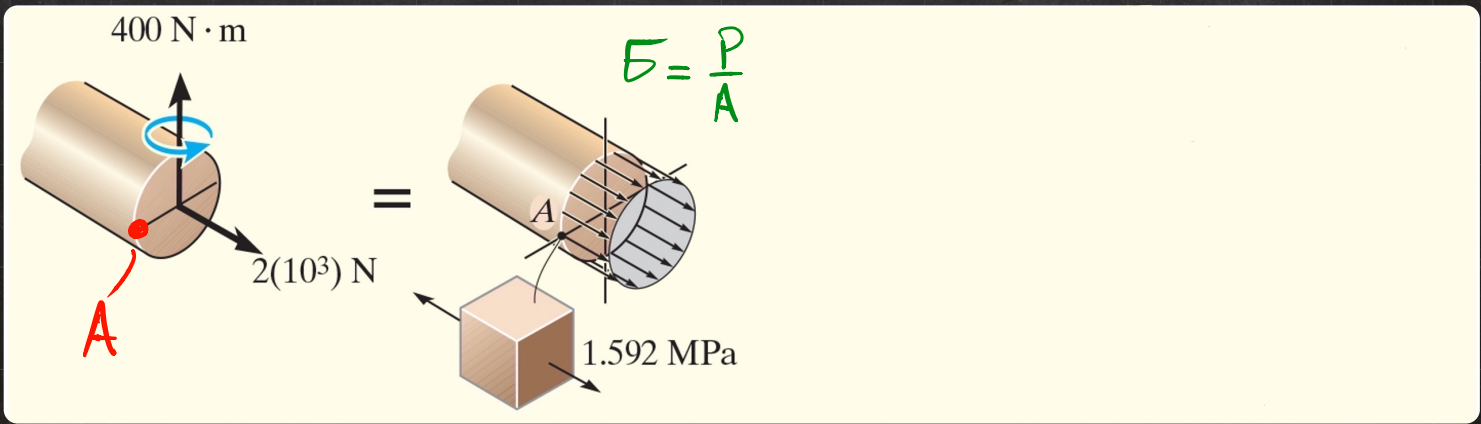
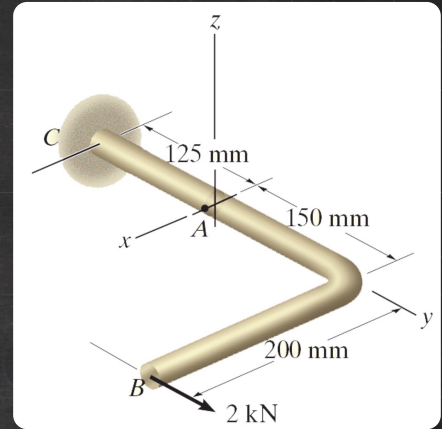
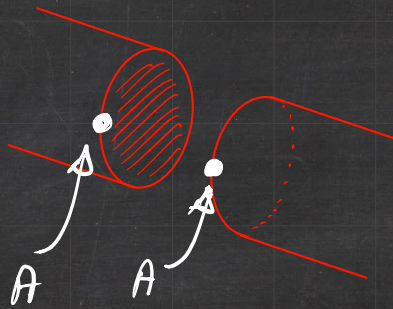
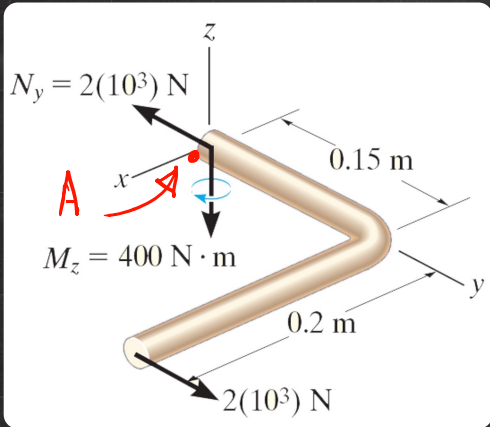


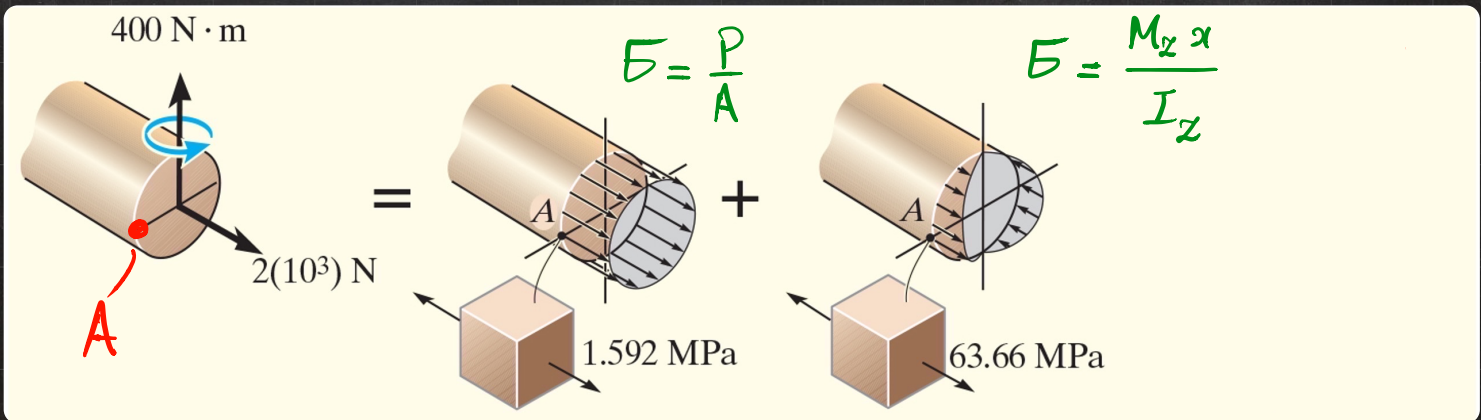
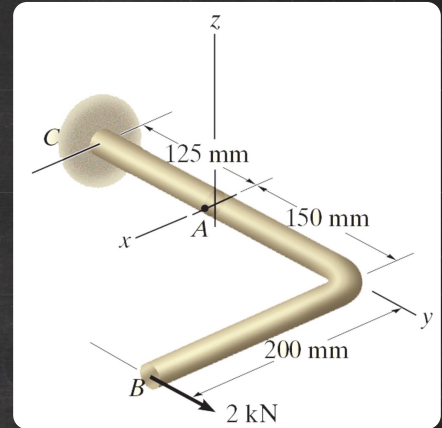
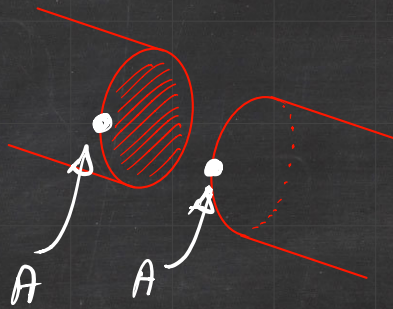
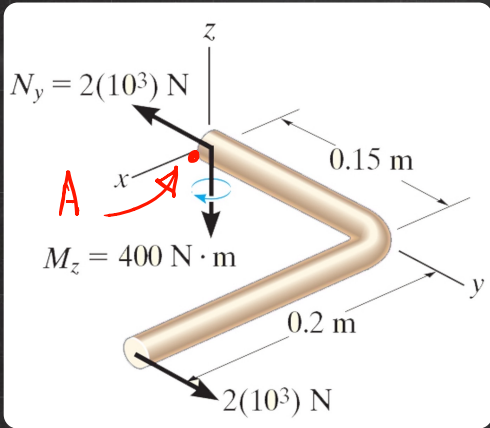


$$\left. \begin{array}{l} \Sigma F_y = 0 \Rightarrow N \checkmark \\ \Sigma M = 0 \Rightarrow M \checkmark \end{array} \right\} \begin{array}{l} \text{CAREFUL} \\ \text{WITH} \\ \text{DIRECTIONS} \end{array}$$

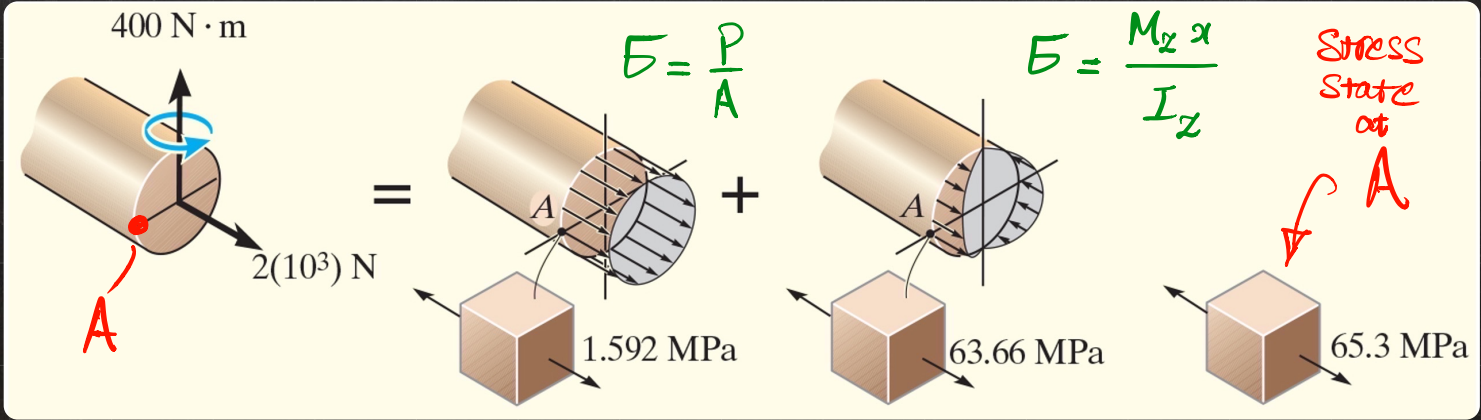
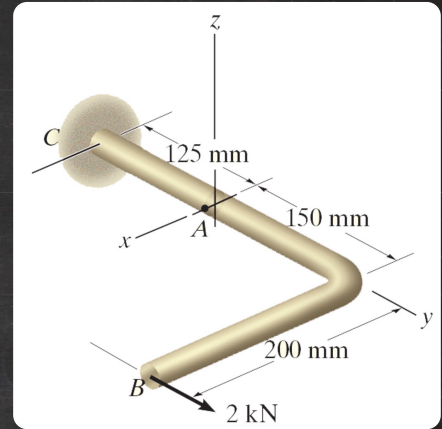
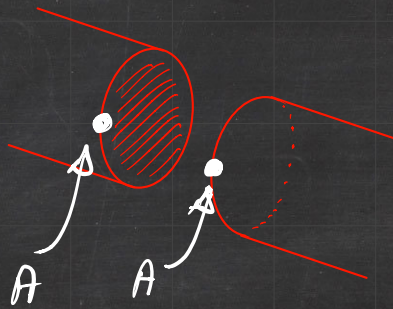
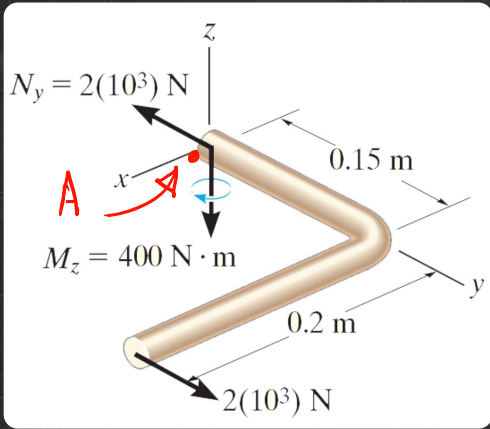






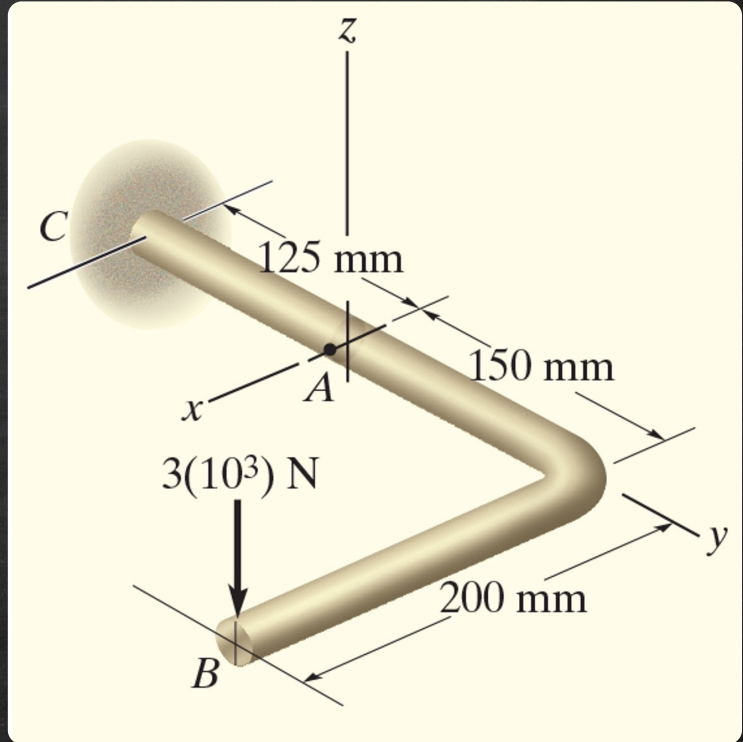






### Exercise 3 . [ similar to ... P. 423 ... 8.6 ]

DETERMINE THE STATE OF STRESS AT POINT A ON THE SOLID ROD SHOWN IN THE FIGURE. THE RADIUS OF THE SHAFT IS 20 mm.



STEP 1)

FIND INTERNAL FORCES/MOMENTS

AT THE DESIRED CROSS-SECTION

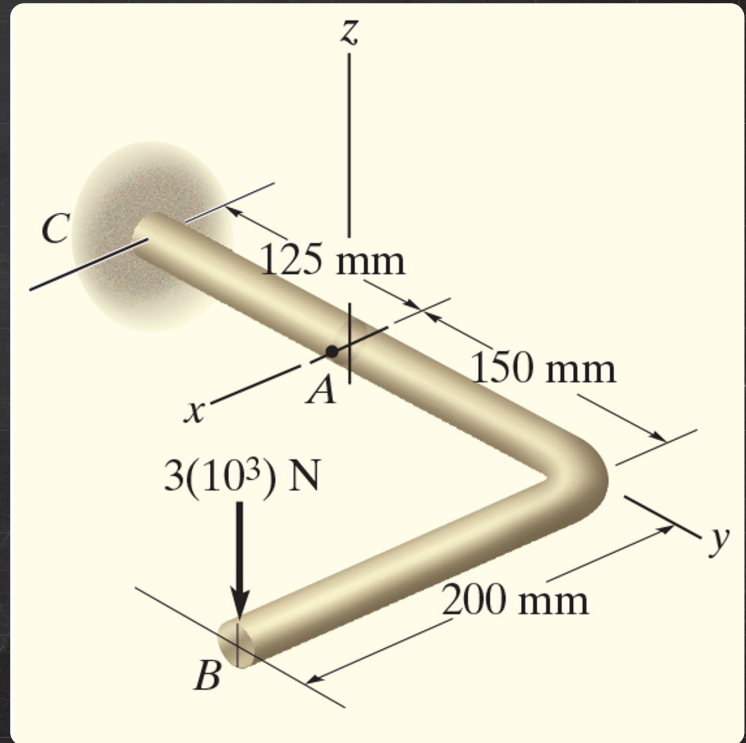
STEP 2)

FIND STRESSES  $\sigma$ ,  $\tau$  CAUSED

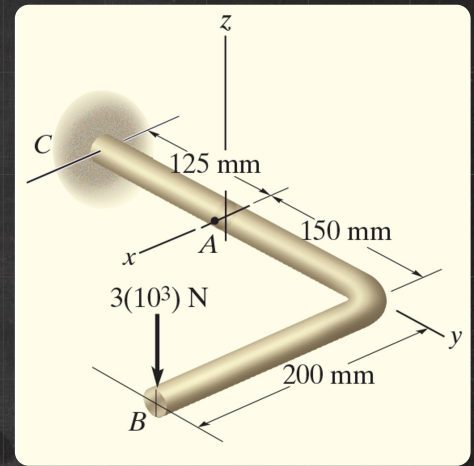
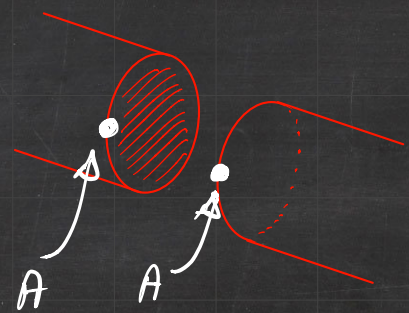
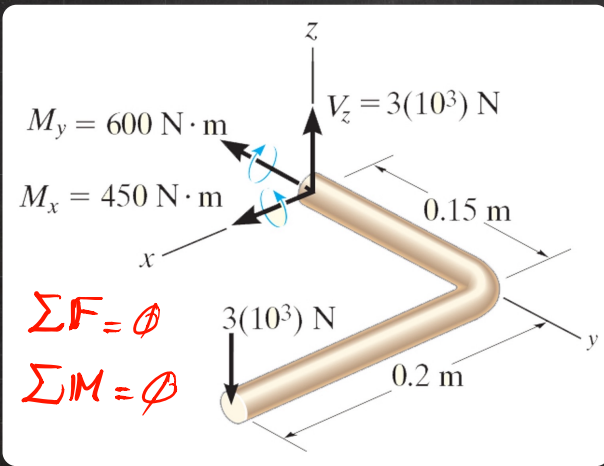
DUE TO INTERNAL REACTIONS

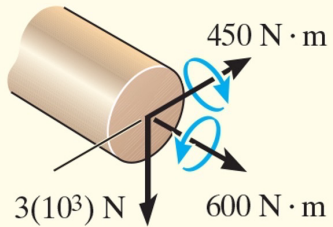
STEP 3)

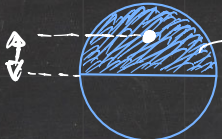
ADD UP THE EFFECTS









$$\bar{y}' = \frac{4c}{3\pi}$$


$$A' = \frac{\pi c^2}{2}$$

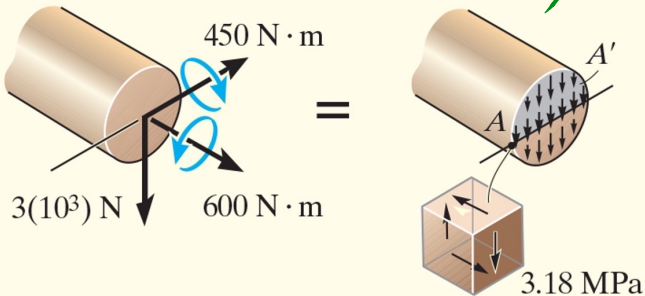
$$Q = \bar{y}' A' = \frac{4c}{3\pi} \left( \frac{1}{2} \pi c^2 \right)$$

thickness  
at A

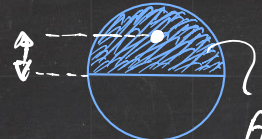
$$t = 2c, \quad I = \frac{\pi}{4} c^4$$

$c = 0.02$

$$\tau = \frac{VQ}{It}$$





$$\bar{y}' = \frac{4c}{3\pi}$$


$$A' = \frac{\pi c^2}{2}$$

$$Q = \bar{y}' A' = \frac{4c}{3\pi} \left( \frac{1}{2} \pi c^2 \right)$$

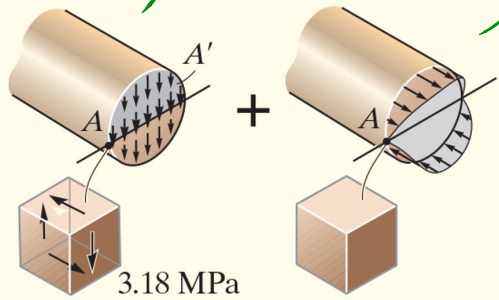
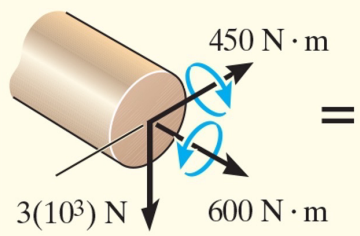
thickness at A  
 $t = 2c$ ,  $I = \frac{\pi}{4} c^4$   
 $c = 0.02$

$$\tau = \frac{VQ}{It}$$

A IS ON NEUTRAL AXIS

$$B_A = 0$$

$$B = \frac{M_x z}{I_x}$$



$$\bar{y}' = \frac{4c}{3\pi}$$


$$A' = \frac{\pi c^2}{2}$$

$$Q = \bar{y}' A' = \frac{4c}{3\pi} \left( \frac{1}{2} \pi c^2 \right)$$

thickness at A

$$t = 2c, \quad I = \frac{\pi}{4} c^4$$

$c = 0.02$

$$\tau = \frac{VQ}{It}$$

A IS ON NEUTRAL AXIS

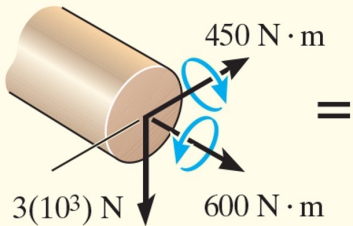
$$B_A = 0$$

$$B = \frac{M_x z}{I_x}$$

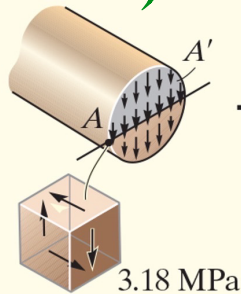
$$T = M_y = 600 \text{ N}\cdot\text{m}$$

$$\tau = \frac{T r}{J}$$

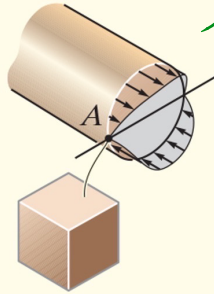
$r = c$   
 $J = \frac{\pi}{2} c^4$



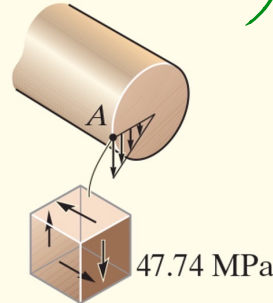
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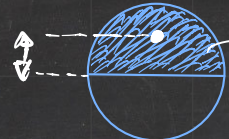


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$$\bar{y}' = \frac{4c}{3\pi}$$


$$A' = \frac{\pi c^2}{2}$$

$$Q = \bar{y}' A' = \frac{4c}{3\pi} \left( \frac{1}{2} \pi c^2 \right)$$

thickness at A

$$t = 2c, \quad I = \frac{\pi}{4} c^4$$

$c = 0.02$

$$\tau = \frac{VQ}{It}$$

A IS ON NEUTRAL AXIS

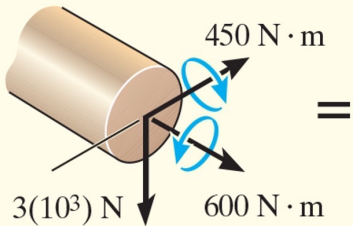
$$B_A = 0$$

$$B = \frac{M_x z}{I_x}$$

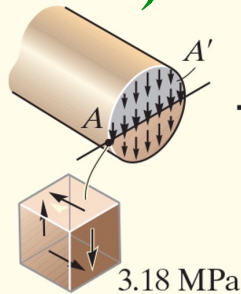
$$T = M_y = 600 \text{ N}\cdot\text{m}$$

$$\tau = \frac{T r}{J} \quad \text{AT A}$$

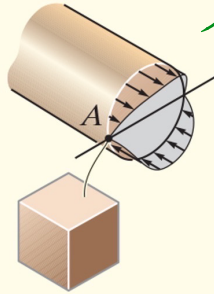
$r = c$   
 $J = \frac{\pi}{2} c^4$



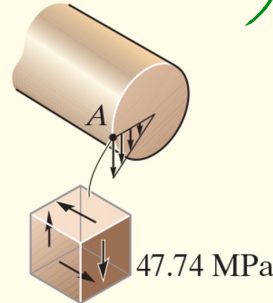
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STRESS STATE AT A





# COMBINED LOADING

\* AXIAL LOADING

$$\sigma = \frac{P}{A}$$

$$\delta = \frac{PL}{EA}$$

\* TORSION

$$\tau = \frac{Tr}{J}$$

$$\theta = \frac{TL}{GJ}$$

\* BENDING

$$\sigma = -\frac{My}{I}$$

$$\theta = \frac{ML}{EI}$$

\* TRANSVERSE SHEAR

$$\tau = \frac{VQ}{It}$$

STEP 1)

FIND INTERNAL FORCES/MOMENTS

AT THE DESIRED CROSS-SECTION

STEP 2)

FIND STRESSES  $\sigma$ ,  $\tau$  CAUSED

DUE TO INTERNAL REACTIONS

STEP 3)

ADD UP THE EFFECTS

(Careful with signs!)

# MECHANICS AND MATERIALS I

MECHANICS AND MATERIALS I

## Combined Loading

Section ... 8.2

Chap. 8

[ Hibbeler 9th edition ]

# MECHANICS AND MATERIALS I

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